### TECHNICAL MANUAL

# UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

This copy is a reprint which includes current pages from Changes 1 through 20.

TM 55-1520-210-PM C20

**CHANGE** 

NO. 20

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 25 October 1999

### **TECHICAL MANUAL**

# UH-1H/V and EH-1H Aircraft PHASED MAINTENANCE CHECKLIST

DISTRIBUTION STATEMENT A Approved for public release; distribution is unlimited

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |  |  |
|---------------|---------------|--|--|
|               | A/(B blank)   |  |  |
| 1-3 and 1-4   | 1-3 and 1-4   |  |  |
| 2-31 and 2-32 | 2-31 and 2-32 |  |  |
| 2-35 and 2-36 | 2-35 and 2-36 |  |  |
| 2-59 and 2-60 | 2-59 and 2-60 |  |  |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

Official:

Joel & Huland JOEL B. HUDSON Administrative Assistant to the Secretary of the Army

9928602

ERIC K. SHINSEKI General, United States Army Chief of Staff

### Distribution:

To be distributed in accordance with Initial Distribution Number (IDN) 310897 requirements for TM 55-1520-210-PM.

TM 55-1520-210-PM C19

CHANGE

NO. 19

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 13 January 1997

### UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited.

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

 Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages      | Insert pages      |  |  |
|-------------------|-------------------|--|--|
| 2-21 and 2-22     | 2-21 and 2-22     |  |  |
| 2-37 and 2-38     | 2-37 and 2-38     |  |  |
| 2-61/(2-62 blank) | 2-61/(2-62 blank) |  |  |

Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

Official

JOEL B. HUDSON Administrative Assistant to the Secretary of the Army 02824 DENNIS J. REIMER General, United States Army Chief of Staff

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31-, block no.0897, requirements for TM 55-1520-210-PM

**CHANGE** 

NO. 18

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., .1 July 1996

## UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited.

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as Indicated below. New or changed text material is Indicated by a vertical bar in the margin. An illustration change Is Indicated by a miniature pointing hand.

| Remove pages       | Insert pages          |  |
|--------------------|-----------------------|--|
| 1-1 through 1-4    | 1-1 through 1-4       |  |
| 1-9 and 1-10       | 1-9 and 1-10          |  |
|                    | 1-10.1/(1-10.2 blank) |  |
| 1-13 and 1-14      | 1-13 and 1-14         |  |
| 2-3 and 2-4        | 2-3 and 2-4           |  |
| 2-11 and 2-12      | 2-11 and 2-12         |  |
| 2-19/(2-20 blank)  | 2-19/(2-20 blank)     |  |
| 2-21 through 2-24  | 2-21 through 2-24     |  |
| 2-29 through 2-32  | 2-29 through 2-32     |  |
| 2-33/(2-34 blank)  | 2-33 and 2-34         |  |
| 2-35 through 2-38  | 2-35 through 2-38     |  |
| 2-49/(2-50 blank)  | 249/(2-50 blank)      |  |
| 2-51 and 2-52      | 2-51 and 2-52         |  |
| 2-59/( 2-60 blank) | 2-59 and 260          |  |
| 2-61/(262 blank)   | 261(262 blank)        |  |

2. Retain this sheet in front of manual for reference purposes.

| TΜ | 55-1520-210-PM |
|----|----------------|
|    | C18            |

By Order of the Secretary of the Army:

DENNIS J. REIMER General, United States Army Chief of Staff

Official:

JOEL B. HUDSON Administrative Assistant to the Secretary of the Army 01977

### **DISTRIBUTION:**

To be distributed in accordance with DA Form 12-31-E, block no. 0897, requirements for TM 55-1520-210-PM

TM 55-1520-210-PM C17

**CHANGE** 

NO. 17

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 15 NOVEMBER 1995

# UH-1H/V and EH-1H Aircraft PHASED MAINTENANCE CHECKLIST

DISTRIBUTION STATEMENT A Approved for public release; distribution is unlimited

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | insert pages  |  |  |
|---------------|---------------|--|--|
| 2-3 and 2-4   | 2-3 and 2-4   |  |  |
| 2-31 and 2-32 | 2-31 and 2-32 |  |  |
| 2-37 and 2-38 | 2-37 and 2-38 |  |  |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

DENNIS J. REIMER General, United States Army Chief of Staff

Oniciai:

YVONNE M. HARRISON
Administrative Assistant to the
Secretary of the Army

### **DISTRIBUTION:**

To be distributed in accordance with DA Form 12-31-E, block no. 0897, requirements for TM 55-1520-210-PM.

CHANGE NO. 16

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 30 April 1992

## UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |  |  |
|---------------|---------------|--|--|
| 1-13 and 1-14 | 1-13 and 1-14 |  |  |
| 2-3 and 2-4   | 2-3 and 2-4   |  |  |
| 2-9/2-10      | 2-9/2-10      |  |  |
| 2-27/2-28     | 2-27/2-28     |  |  |
| 2-29 and 2-30 | 2-29 and 2-30 |  |  |
| 2-31 and 2-32 | 2-31 and 2-32 |  |  |
| 2-51 and 2-52 | 2-51 and 2-52 |  |  |
| 2-53/2-54     | 2-53/2-54     |  |  |
| 2-61/2-62     | 2-61/2-62     |  |  |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

GORDON R. SULLIVAN General, United States Army Chief of Staff

Official: Mitto A. Samelto

MILTON H. HAMILTON
Administrative Assistant to the
Secretary of the Army
01255

### **DISTRIBUTION:**

To be distributed in accordance with DA Form 12-31-E, block no. 0897, PM maintenance requirements for TM 55-1520-210-PM.

CHANGE NO. 15

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 February 1991

UH-1H/V and EH-1H/X Aircraft Phased Maintenance Checklist

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

1-1 through 1-4 2-21 and 2-22 1-1 through 1-4 2-21 and 2-22

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

THOMAS F. SIKORA
Brigadier General, United States Army
The Adjutant General

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31E, block no. 0897, PM requirements for TM 55-1520-210-PM.

TM 55-1520-210-PM C 14

CHANGE )

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 6 September 1990

UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages                | Insert pages                |
|-----------------------------|-----------------------------|
| 1-1 and 1-2                 | 1-1 and 1-2                 |
| 2-3 and 2-4                 | 2-3 and $2-4$               |
| 2-15 through 2-32           | 2-15 through 2-32           |
| 2-35 through 2-42           | 2-35 through 2-42           |
| 2-46.1/2-46.2               | 2-46.1/2-46.2               |
| 2-47 through 2-52           | 2-47 through 2-52           |
| 2-54.1/2-54.2               | 2-54.1/2-54.2               |
| 2-57/2-58 through 2-61/2-62 | 2-57/2-58 through 2-61/2-62 |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

THOMAS F. SIKORA Brigadier General, United States Army The Adjutant General

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM requirements for UH-1H/V Helicopter, Utility, EH-1H Helicopter, Electronic Countermeasure, and EH-1X Helicopter, Electronic Countermeasure & Intercept.

TM 55-1520-210-PM C 13

CHANGE NO. 13

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 16 April 1990

### UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |
|---------------|---------------|
| 2-23/2-24     | 2-23 and 2-24 |
| 2-27/2-28     | 2-27/2-28     |
| 2-31 and 2-32 | 2-31 and 2-32 |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO
General, United States Army
Chief of Staff

Official:

WILLIAM J. MEEHAN II Brigadier General, United States Army The Adjutant General

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM Maintenance requirements for UH-1H/V Helicopter, Utility, EH-1H Helicopter, Electronic Countermeasure and EH-1X Helicopter, Electronic Countermeasure & Intercept.

CHANGE NO. 12

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 November 1989

# UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |
|---------------|---------------|
| 1-3 and 1-4   | 1-3 and 1-4   |
| 2-7/2-8       | 2-7/2-8       |
| 2-23/2-24     | 2-23/2-24     |
| 2-27/2-28     | 2-27/2-28     |
| 2-31 and 2-32 | 2-31 and 2-32 |
| 2-49/2-50     | 2-49/2-50     |
| 2-55 and 2-56 | 2-55 and 2-56 |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

WILLIAM J. MEEHAN II Brigadier General, United States Army The Adjutant General

#### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM Maintenance requirements for UH-1H/V Helicopter, Utility, EH-1H Helicopter, Electronic Countermeasure and EH-1X Helicopter, Electronic Countermeasure & Intercept.

TM 55-1520-210-PM C 11

CHANGE NO. 11

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 August 1988

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages               | Insert pages               |
|----------------------------|----------------------------|
| 2-17 and 2-18<br>2-27/2-28 | 2-17 and 2-18<br>2-27/2-28 |
| 2-31 and 2-32              | 2-31 and 2-32              |
| 2-46.1/2-46.2              | 2-46.1/2-46.2              |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

R. L. DILWORTH
Brigadier General, United States Army
The Adjutant General

#### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM Maintenance requirements for UH-1H/V Helicopter, Utility and EH-1H/X Helicopter, Electronic Countermeasure & Intercept.

### **URGENT**

CHANGE NO. 10

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 6 June 1988

## UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |
|---------------|---------------|
| 1-9 and 1-10  | 1-9 and 1-10  |
| 2-3 and 2-4   | 2-3 and $2-4$ |
| 2-7/2-8       | 2-7/2-8       |
| 2-13/2-14     | 2-13/2-14     |
| 2-17 and 2-18 | 2-17 and 2-18 |
| 2-45/2-46     | 2-45/2-46     |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO
General, United States Army
Chief of Staff

Official:

R. L. DILWORTH
Brigadier General, United States Army
The Adjutant General

#### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM requirements for UH-lH Helicopter, Utility; UH-lV Helicopter, Utility; EH-lH Helicopter, Electronic Countermeasure and EH-lX Helicopter, Electronic Countermeasure and Intercept.

TM 55-1520-210-PM C 9

CHANGE NO. 9

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 26 February 1986

## UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages  | Insert pages  |
|---------------|---------------|
| 1-3 and 1-4   | 1-3 and 1-4   |
| 1-9 and 1-10  | 1-9 and 1-10  |
| 1-13 and 1-14 | 1-13 and 1-14 |
| 2-3 and 2-4   | 2-3 and 2-4   |
| 2-46.1/2-46.2 | 2-46.1/2-46.2 |
| 2-47 and 2-48 | 2-47 and 2-48 |
| 2-51 and 2-52 | 2-51 and 2-52 |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

JOHN A. WICKHAM, JR. General, United States Army Chief of Staff

Official:

MILDRED E. HEDBERG

Brigadier General, United States Army
The Adjutant General

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, AVUM and AVIM Maintenance requirements for UH-1H/V Helicopter, Utility and EH-1H/X Helicopter, Electronic Countermeasure & Intercept.

### **URGENT**

TM 55-1520-210-PM C 8

CHANGE NO. 8

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 15 October 1985

# UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

| Remove pages   | Insert pages   |  |  |
|--|--|--|--|
| 1-15 and 1-16<br>2-35 and 2-36<br>2-41 and 2-42<br>2-53 and 2-54 | 1-15 and 1-16<br>2-35 and 2-36<br>2-41 and 2-42<br>2-53/2-54 |  |  |
| 2-59/2-60<br>2-61/2-62   | 2-54.1/2-54.2<br>2-59/2-60<br>2-61/2-62                      |  |  |

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

JOHN A. WICKHAM, JR. General, United States Army Chief of Staff

Official:

MILDRED E. HEDBERG
Brigadier General, United States Army
The Adjutant General

### DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM Maintenance requirements for UH-1H/V Helicopter, Utility and EH-1H/X Helicopter, Electronic Countermeasure & Intercept.

### **URGENT**

### LIST OF EFFECTIVE PAGES

Insert latest changed pages. Dispose of superseded pages in accordance with regulations.

**NOTE**: On a changed page, the portion of text affected by the latest change is indicated by vertical line in the outer margin of the page. Changes to illustrations are indicated by a miniature pointing hand.

### DATES OF ISSUE FOR ORIGINAL AND CHANGED PAGES ARE AS FOLLOWS:

| Original04 January 1983 | Change1122 August 1988    |
|-------------------------|---------------------------|
| Change131 January 1983  | Change12 November 1989    |
| Change218 March 1983    | Change1316 April 1990     |
| Change318 April 1983    | Change1406 September 1990 |
| Change4. March 1984     | Change1522 February 1991  |
| Change518 April 1985    | Change1630 April 1992     |
| Change619 April 1985    | Change1715 November 1995  |
| Change712 July 1985     | Change1801 July 1996      |
| Change815 October 1985  | Change1913 January 1997   |
| Change926 February 1986 | Change2025 October 1999   |
| Change 10 06 June 1988  |                           |

# TOTAL NUMBER OF PAGES IN THIS PUBLICATION IS 96, CONSISTING OF THE FOLLOWING:

| Page<br>Number | * Change<br>Number | Page<br>Number     | * Change<br>Number | Page '          | Change<br>Number |
|----------------|--------------------|--------------------|--------------------|-----------------|------------------|
| Cover          |                    | 2-11<br>2-12       |                    | 2-402-41 - 2-42 |                  |
| B blank        |                    | 2-12               |                    | 2043            |                  |
| 1-1            | = -                | 2-13<br>2-14 blank |                    | 2-44 blank      |                  |
| 1-2 - 1-3      |                    | 2-14 Diank         |                    | 2-452-45        |                  |
| 1-4            |                    | 2-19               |                    | 2-46 blank      |                  |
| 1-5 - 1-8      |                    | 2-20 blank         |                    | 2-46.1          |                  |
| 1-9            |                    | 2-21               |                    | 2-46.2 blank    |                  |
| 1-10           |                    | 2-22               |                    | 2-47 - 2-48     |                  |
| 1-10.1         |                    | 2-23               |                    | 2-49            |                  |
| 1-10-2         |                    | 2-24               |                    | 2-50 blank      |                  |
| 1-11 - 1-12    | 18                 | 2-25 - 2-26        |                    | 2-51            |                  |
| 1-13 - 1-14    | 18                 | 2-27               |                    | 2-52 - 2-53     | 16               |
| 1-15 - 1-16    | 8                  | 2-28 blank         | 0                  | 2-54 blank      | 16               |
| 2-1            | 0                  | 2-29               | 18                 | 2-54.1          | 14               |
| 2-2 blank      | 0                  | 2-30               | 14                 | 2-54.2 blank    | 0                |
| 2-3            | 17                 | 2-31               | 18                 | 2-55            | 12               |
| 2-4            | 18                 | 2-32               | 20                 | 2-56            | 5                |
| 2-5            | 0                  | 2-33 - 2-34        |                    | 2-57            | 14               |
| 2-6 blank      | 0                  | 2-35               | 8                  | 2-58 blank      | 0                |
| 2-7            | 12                 | 2-36               | 20                 | 2-59            | 20               |
| 2-8 blank      | 0                  | 2-37               | 14                 | 2-60            | 18               |
| 2-9            | 16                 | 2-38               | 19                 | 2-61            | 19               |
| 2-10 blank     | 0                  | 2-39               | 0                  | 2-62 blank      | 0                |

<sup>\*</sup> A zero in this column indicates an original page.

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 4 January 1983

# UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

#### WARNING

CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT THE SPECIFIED INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS ARE PRINTED IN BOLD FACE TYPE.

#### NOTE

INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING PHASED MAINTENANCE AND MUST BE PERFORMED. THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

<sup>\*</sup> This manual supersedes TM 55-1520-210-PM, 20 July 1978, including all changes.

### **SECTION I - GENERAL INFORMATION**

- **1-1. PHASED SCHEDULE.** This phased maintenance inspection checklist contains requirements for inspection of the UH-1H/V and EH-1H/X aircraft on phased schedule having a 900 hour (flight hours) cycle with 150 hour phases. Each requirement included herein is designated for accomplishment at least once, but not more than six times during the 900 hour cycle.
- **EXCEEDING THE PHASED SCHEDULE.** The phased maintenance 1-2. inspection intervals designated are the maximum and shall not be exceeded except in actual operational emergencies as explained herein. It is the Commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat or conditions of disaster which necessitate flight to evacuate aircraft or personnel. Those inspections annotated by a C in the Inspect Phase No's column along with the DA Form 2408-18 (Equipment Inspection Record) items that are due are considered the MINIMUM mandatory Combat maintenance inspection requirements for helicopters scheduled for imminent deployment to or stationed in a combat environment. Under no circumstances will two Combat Phase inspection be performed sequentially. When inspections are delayed to meet emergency requirements, Commanders will assure that the I aircraft status symbol reverts to a red "X" and that delayed inspections are accomplished immediately upon termination of the actual emergency. When unusual local conditions (utilization, type of mission personnel, periods of inactivity, environmental conditions, etc.) dictate, it is the prerogative and responsibility of the Maintenance Officer to increase the scope and/or frequency I of maintenance as necessary to insure safe operation (TM 1-1500-328-23).
- **1-3. MAINTENANCE ACTIVITIES**. The inspections prescribed by this checklist will be accomplished at specified phases by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot Maintenance activities when required.
- **1-4. LIMITATIONS**. The checklist does not contain instructions for repair, adjustment or other means of rectifying conditions. Neither does it contain special tolerances, limits or instructions for special troubleshooting to find causes for malfunctions. Such data will be obtained from the latest issue of the aircraft (TM 55-1520-210-23) series maintenance manuals.
- **1-5. CHANGEOVER TO THE PHASED MAINTENANCE SYSTEM.** Changeover shall be accomplished in accordance with instructions provided in TB 55-1500-337-24 entitled, "Phased Maintenance System for Army Aircraft". The requirements of this TB must be accomplished prior to implementation of Phase 1 inspection requirements specified in this checklist.
- **1-6. PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF).** A preinspection MTF to duplicate nonhazardous equipment problems, determine unsatisfactory conditions, determine equipment operations problems, etc., is recommended prior to start of aircraft disassembly for phased maintenance inspection. The decision to perform the pre-inspection MTF, however, shall be the responsibility of the unit Maintenance Officer.

- 1-7. SPECIAL INSPECTION, CALENDAR INSPECTION AND LUBRICATION REQUIREMENTS. Special inspection, calendar inspection and lubrication requirements contained in C(M 55-1520-210-23) and those listed on the aircraft DA Form 2408-18 shall be reviewed and accomplished in accordance with the "inspection due" requirements specified in those documents.
- **1-8. TIME BETWEEN OVERHAUL (TBO) AND RETIREMENT LIFE ITEMS CHECK.** Prior to start of the applicable phased maintenance inspection, a check will be made of components and their remaining operating hours prior to removal. The latest issue of the aircraft, TM 55-1520-210-23 and DA Form 2408-16, shall be referred to for a complete listing of components and their TBO and retirement life.
- **1-9. USING THE PHASED INSPECTION CHECKLIST**. For use of the phased inspection checklist, refer to DA Pam 738-751.
- 1-10. FINAL RECORDS CHECK. After all corrective actions have been completed and following completion of the phased inspection, the technical inspector or designated supervisor shall verify that all applicable forms and records have been properly updated. Any fault not corrected will be carried forward to a new DA Form 2408-13 or reentered on DA Form 2408-14. A final records checklist (Table 1-2) is provided to ensure forms and records have been inspected for completeness and accuracy prior to release of the aircraft from the phased maintenance inspection. The inspector verifying the final records check shall enter his initials adjacent to the indicated form or record on the Final Records Checklist. The initials entered shall be registered on the Signature Sheet (Table 1-1) adjacent to that person's signature.

- **1-11. SIGNATURE SHEET.** All personnel performing inspection and/or maintenance tasks shall place their signatures and initials on the signature sheet (Table 1-1). The purpose of the signature sheet is to provide a correlation between initials entered on the individual checklist sheets and the actual names of the personnel accomplishing these tasks.
- **1-12. MAINTENANCE OPERATIONAL CHECKS.** After the completion of any required corrective actions to any of the components of the function system of the aircraft, maintenance operational checks (MOC) shall be performed on that system to determine the effectiveness of maintenance actions performed and to verify to proper operation of that system. These MOC shall be performed in accordance with TM 1-1500-328-23. Copies of DA Form 240813-1 (Figure 1-1) and DA Form 2408-13-2 (Figure 1-1.1) may be used to record and sign off the MOC performed.
- **1-13. MAINTENANCE TEST FLIGHT**. When all required inspections in Section II have been accomplished and initialed in accordance with the above procedures, a daily inspection in accordance with the TM specified in Section II will be preformed on the aircraft to permit a maintenance test flight (MTF) to be made. The MTF shall be performed in accordance with the requirements of TM 55-1520-242-MTF and TM 1-1500-328-23, using the MTF Form in the MTF Technical Manuals. A suggested maintenance test flight checksheet (Figure 1-5) and Rotor Smoothing Record (Figure 1-6) are provided at the end of Section I.
- **1-14. CHECKLIST DISPOSITION.** The completion of each phased maintenance shall be recorded on DA Form 2408-13 and DA Form 2408-15 as prescribed by DA PAM 738-751. The signed checklist, together with all continuation sheets shall be attached to DA Form 2408-13, and filed for the six month period as required by DA PAM 738-751.

- **1-15. INSPECTION AREAS.** Figures 1-2 and 1-3 show the inspection areas of the UH-1H/V and EH-1H/X aircraft. These areas are titled as shown below. Figure 1-4 shows the location of access doors and panels which require removal at various phased maintenance inspections. Access panels and doors are identified by number of the text where applicable. Additional panels and doors may be removed as required to facilitate inspection requirements.
- **1-15.1** This TM checklist may contain inspection requirements applicable to specific equipment not installed on your aircraft. Those requirements should be disregarded

- **1-16**. Deleted.
- 1-17. REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS. You can help improve this manual. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms), or DA Form 2028-2 located in the back of this manual direct to: Commander, U.S. Army Aviation and Missile Command (AMCOM), ATTN: AMSAM-MMC-LS-LP, Redstone Arsenal, AL 35898-5230. You may also submit your recommended changes by E-mail directly to Is-lp@redstone.army.mil, or by fax at 256-842-6546 or DSN 788-6546. A reply will be furnished directly to you. Instructions for sending an electronic 2028 may be found at the back of this manual immediately preceding the hardcopy DA Forms 2028.

| AREA NO. | AREA TITLE                            |
|----------|---------------------------------------|
| 1        | Aircraft Exterior                     |
| 2        | Nose Area                             |
| 3        | Forward Radio/Battery Compartment     |
| 4        | Cockpit Interior                      |
| 5        | Cabin Interior                        |
| 6        | Under Floor Cockpit/Cabin             |
| 7        | Lower Pylon Area (Via Cabin Interior) |
| 8        | Upper Pylon Area (Via Cabin Roof)     |
| 9        | Main Rotor and Mast Area              |
| 10       | Engine Air Induction Area             |
| 11       | Engine Compartment                    |
| 12       | Cabin Roof                            |
| 13       | Cabin Sides, Bottom and Landing Gear  |
| 14       | Under Cabin Pylon Area (Hell Hole)    |
| 15       | Mid Fuselage Under Engine Deck        |
| 16       | Electronic Comm. Compartments         |
| 17       | Engine Area Exterior                  |
| 18       | Tailboom Interior                     |
| 19       | Tail Rotor Drive Train Area           |
| 20       | Tail Rotor and Gearbox Area           |
| 21       | Oil Cooler Aft Battery Compartment    |
| 22       | Heater Compartment                    |

### Table 1-1. Signature Sheet (Sheet 1 of 3)

| Signature of Person Accomplishing Necessary Work | Initial |
|--|---------|
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Maintenance Supervisor              | Initial |
| Signature of Technical Inspector                 | Initial |
| Signature of Maintenance Officer                 | Initial |

### Table 1-1. Signature Sheet (Sheet 2 of 3)

| Signature of Person Accomplishing Necessary Work | Initial |
|--|---------|
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
| Signature of Person Accomplishing Necessary Work | Initial |
|  |         |
| Signature of Maintenance Supervisor              | Initial |
| Signature of Technical Inspector                 | Initial |
| Signature of Maintenance Officer                 | Initial |

# Table 1-1. Signature Sheet (Sheet 3 of 3)

| Signature of Person Accomplishing Necessary Work | Initial         |
|--|-----------------|
| Signature of Person Accomplishing Necessary Work | Initial         |
| Signature of Person Accomplishing Necessary Work | Initial         |
| Signature of Person Accomplishing Necessary Work | Initial         |
| Signature of Person Accomplishing Necessary Work | Initial         |
| Signature of Person Accomplishing Necessary Work | Initial         |
| Signature of Maintenance Supervisor              | Initial         |
| Signature of Technical Inspector                 | Initial Initial |
| Signature of Maintenance Officer                 |                 |

#### Table 1-2. Final Records Checklist

This checklist is provided to insure the indicated forms and records have been inspected for presence, completeness, legibility, and accuracy prior to releasing the aircraft from a phase inspection. Vertification of inspection will be indicated by placing the initials of the inspector in the appropriate initial block (refer to DA PAM 738-751).

| DA FORM 2408-5         |  |
|------------------------|--|
| <br>Delete             |  |
| DA FORM 2408-15        |  |
| <br>DA FORM 2408-16    |  |
| <br>DA FORM 2408-17    |  |
| DA FORM 2408-19        |  |
| DA FORM 2408-20        |  |
| LOCALLY REQUIRED FORMS |  |
|                        | Delete DA FORM 2408-15 DA FORM 2408-16 DA FORM 2408-17 DA FORM 2408-19 DA FORM 2408-20 |

| PRODUCTION CONTROL | 15.1171.51 | CHALITY CONTROL        | INITIAL |
|--------------------|------------|------------------------|---------|
| RECORDS            | INITIAL    | QUALITY CONTROL        | HALLIAL |
| FLOW CHART         |            | TBO FILE               |         |
| STATUS BOARD       |            | QA FILE                |         |
| WORK ORDER FILE    |            | SERIAL NUMBER FILE     |         |
| MWO FILE           |            | AOAP FILE              |         |
| Delete             |            | INVENTORY RECORDS      |         |
| 1352 REPORTS       |            | WEIGHT AND BALANCE     |         |
| LOCAL RECORDS      |            | MSG FILE               |         |
| <u> </u>           |            | DA FORM 2410 SUBMITTED |         |
|                    |            | LOCAL RECORDS          |         |

| 1. AF                                | AIRCRAFT SERIAL NUMBER 2. MODEL       |              |                |             |         |       |          | 3. DATE 4. PAGE |              |        |                                  |          |     |               |              |                |     |          |
|--------------------------------------|---------------------------------------|--------------|----------------|-------------|---------|-------|----------|-----------------|--------------|--------|----------------------------------|----------|-----|---------------|--------------|----------------|-----|----------|
|                                      |                                       |              | P/             | ART I - FAL | LT INFO | RMATI | ON       |                 |              |        | PART II - CORRECTING INFORMATION |          |     |               |              |                |     |          |
|                                      | STATUS                                | <b>5</b> Y9  |                | DATE        | NO.     |       | TIME     |                 | Ţ            | PID    | DATE                             |          |     | TIME          |              |                | HAS |          |
|                                      | <u> </u>                              |              |                |             |         |       |          |                 | $\perp$      |        | PIOUNDS                          |          |     | ACTIO         | ON CODE      |                | WUC |          |
| FAUL                                 | T/REMARK                              | 3            |                |             |         | _     |          |                 |              |        | ACTION                           |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          |                 |              |        |                                  |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          |                 |              |        |                                  |          |     |               | <del>,</del> |                |     |          |
|                                      |                                       |              |                |             |         |       | PID      | HOU             | RS           | PID    | HOURS                            | PID      |     | HOURS         |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          | $\vdash$        | <b>-</b> ∔   |        | 4                                | _        |     |               |              |                |     |          |
| A/C HRS WHEN DISC HOW REC MALEFF WUC |                                       |              |                |             |         |       | CMH      | ι               | DMH          |        | FMH                              | <u> </u> | DMH |               |              |                |     |          |
|                                      |                                       |              | <del>- 1</del> |             | OW NEC  |       |          |                 |              |        | 4                                |          | OWN |               |              |                |     |          |
| W.O.                                 |                                       | <del>,</del> |                | REQ         |         |       | OTHE     |                 |              |        | TIPID                            |          |     | <del></del> - | TI MAN-HO    |                |     |          |
|                                      | STATUS                                | SYS          | ļ              | DATE        | NO.     |       | TIME     |                 |              | PID    | DATE                             |          |     | TIME          |              |                | HRS |          |
|                                      | <u> </u>                              | <u> </u>     |                |             |         |       | L        | <del></del>     | Ц            | _      | ROUNDS ACTION CODE               |          |     | ON CODE       | wuc          |                |     |          |
| FAUL                                 | T/REMARK                              | 8            |                |             |         |       |          |                 |              |        | ACTION                           |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          |                 |              |        |                                  |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          |                 |              |        | PID                              | но       | IDe | PIO           | HOURS        | PID            |     | HOURS    |
|                                      |                                       |              | _              |             |         |       |          |                 |              |        | -                                | -        |     |               | 1.00.10      | <del>  ~</del> |     | 1.000.00 |
|                                      | ·                                     |              |                |             |         |       | -        |                 |              |        |                                  |          |     |               |              | ╁─             |     |          |
| A/C I                                | #RS                                   |              | WHEN           | DISC        | HOW RE  | C     | MAL      | EFF W           | uc           |        | СМН                              | OMH OMH  |     |               | FMH DMH      |                | •   |          |
| W.O.                                 |                                       |              |                | REQ         |         |       | отн      | ER              |              |        | ,TPID                            |          |     |               | TI MAN-H     | MAN-HOURS      |     |          |
|                                      | STATUS                                | SYS          |                | DATE        | NO.     |       | TIME     |                 | T            | PID    | DATE                             |          |     | TIME          |              |                | HRS |          |
|                                      | 13.2.0                                | <u> </u>     |                |             |         |       | <u> </u> |                 |              |        | ROUNDS                           |          |     | ACT           | ON CODE      |                | WUC | ·        |
| FAUL                                 | T/REMARK                              |              |                |             |         |       |          |                 |              |        | ACTION                           |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          |                 |              |        |                                  |          |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       | <u> </u> | ·<br>1:         |              |        | <del></del>                      | 1        |     |               |              |                |     |          |
|                                      |                                       |              |                |             |         |       |          | PID             | HO           | URS    | PID                              | HOURS    | PIO |               | HOURS        |                |     |          |
| <b> </b>                             |                                       |              |                |             |         |       |          | }               | <del> </del> |        |                                  | -        | ┼   |               | <b> </b>     |                |     |          |
| 45                                   | A/C HRS WHEN DISC HOW REC MAL EFF WUC |              |                |             |         |       |          | CIN             |              | لــــا |                                  | 15.4.    |     | T 214.        | L            |                |     |          |
|                                      |                                       |              | AALIEI         |             | I HOW H |       | ₩        | L               |              |        | CMH OMH FMH DMH                  |          |     |               |              |                |     |          |
| W.O. REQ OTHER                       |                                       |              |                |             | TIPID   |       |          |                 | H-MAM IT     | OURS   |                                  |          |     |               |              |                |     |          |

DA FORM 2408-13-1, OCT 91

AIRCRAFT INSPECTION AND MAINTENANCE RECORD For use of this form, see DA PAM 736-751; the proponent agency is DCSLOG

Figure 1-1. DA Form 2408-13-1

|         | Ü           | DATE             | <br>rays        |     |         |         |  |  |  |  |
|---------|-------------|------------------|-----------------|-----|---------|---------|--|--|--|--|
|         |             | 2. SERIAL NUMBER | 3. SYSTEM CODE  |     | 4. TIME |         |  |  |  |  |
| 1. STAT | /Us         | 5. FAULT DATE    | 6, FAULT NUMBER |     | i       |         |  |  |  |  |
| 7. FAU  | LT          |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         | <del></del> |                  |                 |     |         |         |  |  |  |  |
| 8.STA   | 9. FAULT    |                  | <br>IO. ACTION  | 11. | . PIO   | 12. MMH |  |  |  |  |
|         | <b></b>     |                  |                 |     |         |         |  |  |  |  |
|         | 1           |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  | ,               |     |         | _       |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             | _                |                 |     |         |         |  |  |  |  |
|         |             |                  | <br>            |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |
|         |             |                  |                 |     |         |         |  |  |  |  |

DA FORM 2408-13-2, NOV 91

RELATED MAINTENANCE ACTIONS RECORD

For use of this form, see DA PAM 738-751; the proponent agency is DCSLOG

Figure 1-1.1. DA Form 2408-13-2

1-10.1/(1-10.2 blank) C18

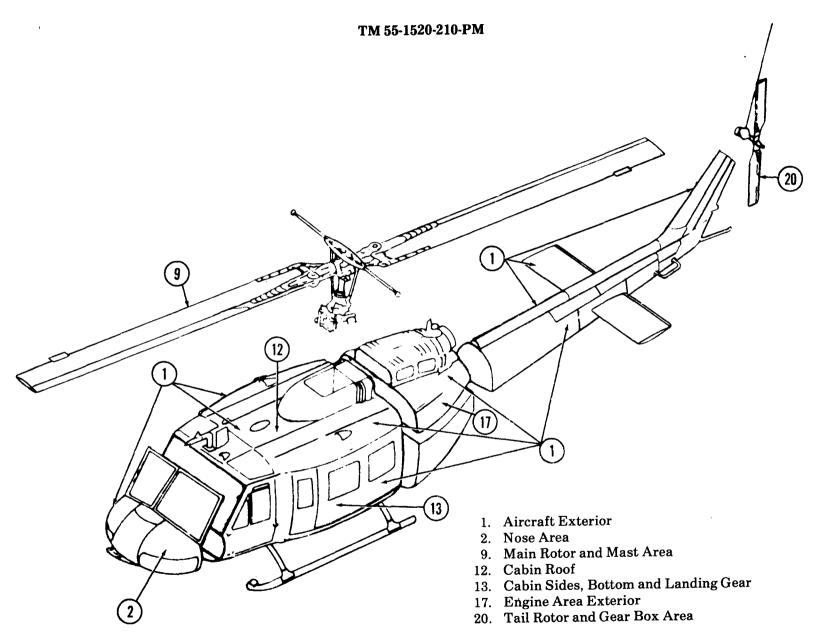


Figure 1-2. Exterior Inspection Areas

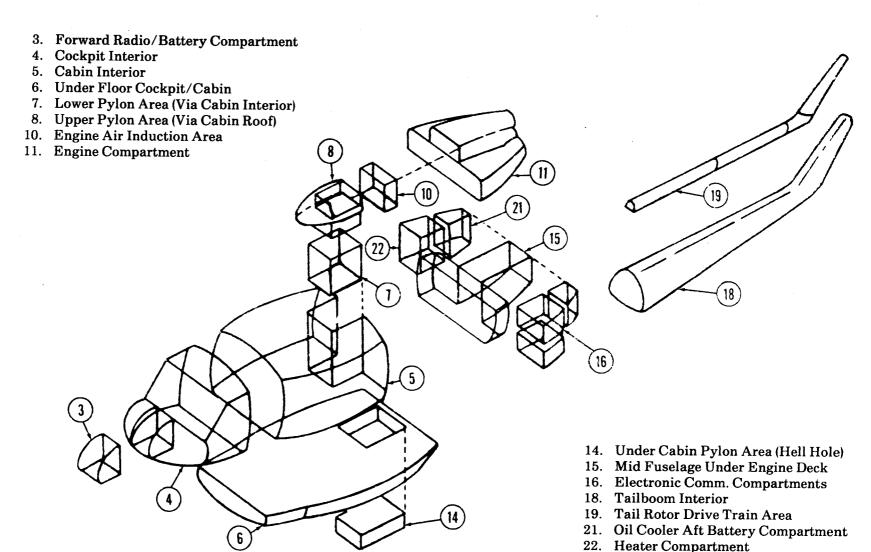


Figure 1-3. Interior Inspection Areas

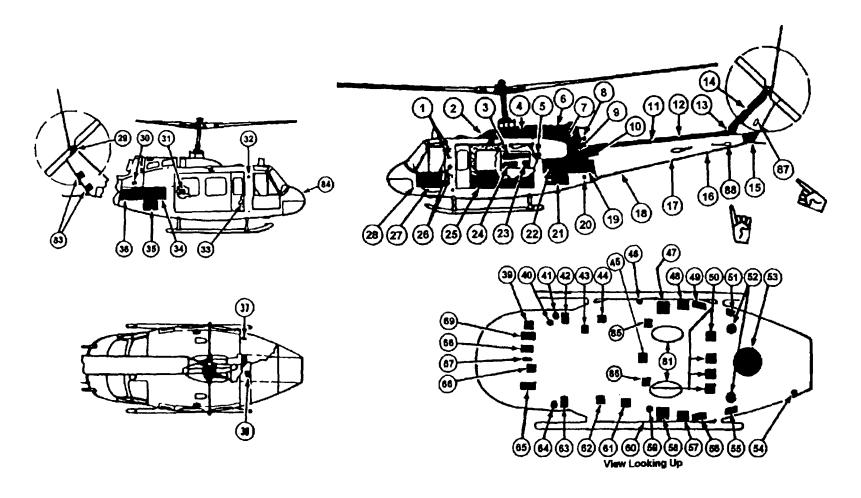


Figure 1-4. Model UH-1H/V and EH-1H/X Access and Inspection Provisions (Sheet 1 of 2)

1-13 C18

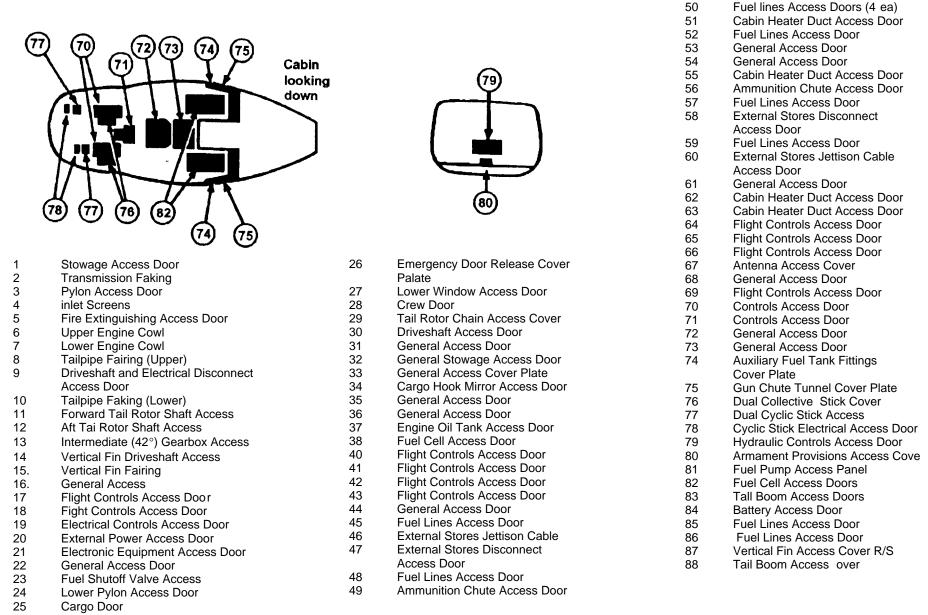


Figure 14. Model UH-1H/V and UH-1H/X Access and Inspection Provisions (Sheet 2 of 2)

C 8

1-15

| UH−1 CE  | H.M.V. & EH-1H/x                                |
|--|---|
| PURPOSE OF MIT FAT   | ACFT MODEL & SN DATE                            |
| PILOT  | UNIT  |
| F ( CO )   |   |
| SYMBOLS V= SATISFACTORY X = DEFI                                     | CIENCY  |
| Prior to Test Flight   | 2. Power Check                                  |
| 1. Forms and Records   | 3. Control Responses Checks                     |
| 2. Flight Readiness Insp.  | 4. Pylon Mounts Checks                          |
| <ol><li>Weight and Balance</li></ol>                                 | 5. Engine Response                              |
| 4. Engine Baseline Data  | 6. Power Cylinder                               |
| TO N1 EGT  | 7. Low RPM Hover                                |
| Starting Engine  | 8. Hover in Emergency 9. Torque MeterPSI        |
| 1. Press to Test Lights  | 9. Torque Meter PSI LEVEL OFF CHECKS            |
| <ol> <li>Fire Warning Light</li> <li>Caution Panel Lights</li> </ol> | 1. Eng Oil PressTemp                            |
| 4. Throttle System Cushion   | 2. XMSN Oil Press Temp                          |
| OPEN CLOSED  | 3. EGT°C  |
| ENGINE RUNUP   | 4. Airspeed Indicators                          |
| l. Engine Idle % Nl  | INFLIGHT CHECK                                  |
| 2. Emerg Gov Switch  | 1. Control Rigging                              |
| 3. Hydraulic System  | 2. Autorotation RPM                             |
| 4. Fuel Boost Pumps  | 3. Hydraulics Off                               |
| Right PSI Left PSI   | 4. TEAC: PA                                     |
| 5. Bleed Band Operation  | FAT TQ  |
| OPEN CLOSE %N1FAT  | N1 EGT  |
| 6. Variable Inlet Gui de Vanes                                       |   |
| BEGIN TO OPEN % N1FAT  | 6. Vibration Analysis                           |
| 7. Fuel Quantity Gage  | 7. Cyclic Rigging                               |
| 8. Pitot Heater  | 8. Fuel Consumption Initiate TIME FUEL          |
| 9. Spare Inverter AB AC BC   | 9. Instruments                                  |
| 10. Main GEN VDC   | Altimeters Att. Ind.                            |
| 11. STBY GENVDC  | VSI Stby Comp                                   |
| 12. Main Inverter  | RMI Clock                                       |
| AB AC BC   | Turn & Slip                                     |
| 13. Bleed Air Heater   | 10. Comm/NAV Radios                             |
| 14. Deice Operation  | UHF VHF FM1 FM2                                 |
| 15. Low RPM Warning  | VOR ADF LOC/GS                                  |
| Off On   | MB XPONDER                                      |
| 16. GOV INCR/DECR  | MODE C  |
| Full Incr DECR   | 11. Fuel Consump. Complete                      |
| Travel Time SEC 17. High RPM Warning                                 | Time Fuel AFTER LANDING/ENGINE SHUTDOWN         |
| OFF ON   | 1. EGT  |
| 18. Force Trim System  | 2. Eng Oil Press Temp                           |
| 19. Collective Friction  | 3. XMSN Oil Press Temp                          |
| UP DOWN  | 4. Eng Idle %N1                                 |
| 20. Eng Oil PressTemp  | 5. Battery                                      |
| 21. XMSN Oil PressTemp   | 6. Eng Oil Press Lite                           |
| 22. Fuel PressPSI  | 7. XMSN Oil Press Lite                          |
| 23. Torque Press PSI   | 8. N1 Coastdown Time Sec                        |
| 24. EGT °C   | 9. Emerg. Collective Accum.                     |
| 25. Altimeters P CP BEFORE TAKEOFF                                   | 10. Post Flight<br>11. Forms & Records Complete |
| 1. HIT Check   | SPECIAL REQUIREMENT (LIST)                      |
| HOVER CHECK  | 1.  |
| 1. Takeoff to hover  | 2.  |
|  |   |

|                      | ROTOR SMOOTHING RECORD |        |         |        |                      |        |              |         |        |  |
|----------------------|------------------------|--------|---------|--------|----------------------|--------|--------------|---------|--------|--|
|                      | BLADE<br>RIAL M        | IUMBER |         |        |                      | TE BL/ | ADE<br>NUMBE | R       |        |  |
| ADJUSTMENT<br>NUMBER | TAB                    | ROLL   | BALANCE | EFFECT | ADJUSTMENT<br>NUMBER | TAB    | ROLL         | BALANCE | EFFECT |  |
| 1                    |                        |        |         |        | 1                    |        |              |         |        |  |
| 2                    |                        |        |         |        | 2                    |        |              |         |        |  |
| 3                    |                        |        |         |        | 3                    |        |              |         |        |  |
| 4                    |                        |        |         |        | 4                    |        |              |         |        |  |
| 5                    |                        |        |         |        | 5                    |        |              | . =     |        |  |
|                      |                        |        |         | REM.   | ARKS                 |        | <del></del>  |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |
|                      |                        |        |         |        |                      |        |              |         |        |  |

Figure 1-6. Suggested Format of Rotor Smoothing Record

PILOIS SIGNATURE

C 8 1.16

# SECTION II - INSPECTION CHECKLIST

#### NOTE

PRIOR TO START OF THE PHASED MAINTENANCE INSPECTION, IT IS RECOMMENDED THAT A PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF) BE CONDUCTED. ACCOMPLISHMENT OF THE MTF SHALL BE DETERMINED BY THE UNIT MAINTENANCE OFFICER. THE PRE-INSPECTION MTF SHOULD BE CONDUCTED BY A MAINTENANCE TEST PILOT FOLLOWING A REVIEW OF THE AIRCRAFT FORMS AND RECORDS AND A BRIEFING FROM THE REGULAR FLIGHT CREW OF THE AIRCRAFT. THE MTF IS RECOMMENDED TO ASSESS THE AIRCRAFT PERFORMANCE AND IDENTIFY DEFICIENCIES THAT SHOULD BE CORRECTED WHILE THE AIRCRAFT IS UNDERGOING PHASED INSPECTION.

| PHASE                    | E No  |        | Area Name and No.<br>GENERAL | Aircraft Serial No. |              |  |         |
|--------------------------|---|--------|------------------------------|---------------------|--------------|--|---------|
| Inspect<br>Phase<br>No's | Inspection Requirements   | Status | Faults and/or Remarks        |                     | Action Taken |  | Initial |
| ALL                      | 1. Prior to inspection, check aircraft forms and records for deficiencies (use Table 1-2 for reference to aircraft forms and records).  |        |                              |                     |              |  |         |
| ALL                      | 2. Clean engine in accordance with TM 55-2840-229-23.   |        |                              |                     |              |  |         |
| ALL                      | 3. Clean aircraft in accordance with the latest issue of the aircraft AVUM and AVIM maintenance manuals.  |        |                              |                     |              |  |         |
|                          | 4. Deleted.   |        |                              |                     |              |  |         |
| ALL                      | 5. Aircraft without ODDS, check all electrical chip detectors (except engine) for metal accumulation, clean, perform functional check and reinstall. Refer to TM 55-2840-229-23 for engine chip detector check. |        |                              |                     |              |  |         |

# "FOD REMINDER"

Check work area for tools and parts after completion on maintenance and inspection.

| Phase                   | Phase No GENERAL (CONT) |  |  | Name and No |                 |         | Aircraft Serial No. | Dat | te      |
|-------------------------|-------------------------|--|--|-------------|-----------------|---------|---------------------|-----|---------|
| Inspect<br>Phase<br>No. |                         | Inspection R   | lequirements   | Status      | Faults and/or l | Remarks | Action Taken        |     | Initial |
| 3,6                     | 6.                      | Defuel aircraft in accordance with TM 55-<br>1520-210-23 prior to removal of floor panels.                       |  |             |                 |         |                     |     |         |
| 1,2,<br>4,5             | 7.                      | of phased inspections  | ully serviced prior to start<br>s. If maintenance is to be<br>required defueling, this<br>ferred until after such<br>leted.  |             |                 |         |                     |     |         |
| ALL<br>C                | 8.                      | electrical equipment avionics publications   | spections, check and test<br>as required in applicable<br>s. Any faults discovered<br>is shall be entered on DA<br>8-13-1-E. |             |                 |         |                     |     |         |
| ALL                     | 9.                      | and test as required   | system inspection checks<br>in applicable armament<br>ults discovered during the<br>entered on DA Form 2408-                 |             |                 | 11-1    |                     |     |         |
| ALL<br>C                | 10.                     | Perform engine ex-<br>functional test in ac<br>manuals. Any fault<br>inspections shall be e<br>13-1/2408-13-1-E. | xhaust gas temperature<br>cordance with applicable<br>s discovered during the<br>entered on DA Form 2408-                    |             |                 |         |                     |     |         |

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

| PHASE NO                 |                    | Area N<br>AIRCRAFT EXTERIO |              | nd No.                                | Aircraft Serial No.                     |              |  |              |
|--------------------------|--------------------|----------------------------|--------------|---------------------------------------|---|--------------|--|--------------|
| Inspect<br>Phase<br>No's | Inspection (       | Requirements               | Status       | Faults and/or R                       | lemarks                                 | Action Taken |  | leifini      |
|                          | NO                 | TE                         |              |                                       |   |              |  |              |
|                          | This page intentio | nally left blank.          |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            | <u> </u>     |                                       |   |              |  |              |
|                          |                    |                            |              |                                       | ·                                       |              |  | <u> </u>     |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  | <del> </del> |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            | <b> </b>     | · · · · · · · · · · · · · · · · · · · |   |              |  |              |
|                          |                    |                            | ļ            |                                       |   |              |  | <u></u>      |
|                          |                    |                            | ļ            |                                       |   |              |  |              |
|                          |                    |                            | <b></b>      |                                       |   |              |  |              |
|                          |                    |                            | <b> </b>     |                                       |   |              |  | <b> </b>     |
|                          |                    |                            |              |                                       | ······································  |              |  | ļ            |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       |   |              |  |              |
|                          |                    |                            |              |                                       | *************************************** |              |  |              |
|                          |                    |                            | <b></b>      |                                       |   |              |  |              |
|                          |                    |                            | <b> </b>     |                                       |   |              |  |              |
|                          |                    |                            | <del> </del> |                                       | <del></del>                             |              |  | ļ            |
|                          |                    |                            |              | <u> </u>                              |   | ł            |  | L            |

| PH                       | ASE NO                                      | Area N<br>NOSE AREA - 2    | ame ar       | nd No.        | A           | rcraft Serial No. | Date                                   | 2  |
|--------------------------|---|----------------------------|--------------|---------------|-------------|-------------------|--|--|
| Inspect<br>Phase<br>No:s | Inspection I                                | Requirements               | Status       | Faults and/or | r Remarks   | Action Taken      |  | Inilial  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  | ļ  |
|                          |   |                            | <u> </u>     |               |             |                   |  | ļ  |
|                          |   |                            | <b></b> -    | <u></u>       |             |                   |  |  |
|                          |   |                            | <del> </del> | <u> </u>      |             |                   |  |  |
|                          |   |                            | <b></b>      |               |             |                   |  |  |
| 2,4,6                    | 2. Wiper blade arms for condition, security |                            | <b>}</b>     |               |             |                   |  | l  |
|                          | and proper adjustment.                      |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
| 3 & 6                    | Remove windshield wip                       | per motor and converter    |              |               |             |                   |  |  |
|                          | assy, disassemble, clean                    | , inspect and service con- | <u> </u>     |               |             |                   |  | ļ  |
|                          | verter.                                     |                            | <b> </b>     |               |             |                   |  | <b>├</b> ──                                      |
|                          |   |                            | <b> </b>     |               | <del></del> |                   |  | <del> </del>                                     |
|                          |   |                            |              |               |             |                   |  | <b></b>  |
|                          |   |                            | <b></b>      |               |             |                   | ······································ | <u> </u>   |
|                          |   |                            | ļ            | <br>          |             |                   |  | <b> </b>   |
|                          |   |                            | <b></b>      |               |             |                   | ·                                      | <b>}</b>   |
|                          |   |                            |              | <u> </u>      | <del></del> |                   |  | <del> </del>                                     |
|                          |   |                            | <b></b>      |               |             |                   |  | <del>                                     </del> |
|                          | <del></del>                                 |                            |              |               |             |                   |  | <del></del>                                      |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            |              |               |             |                   |  |  |
|                          |   |                            | 1            |               |             |                   |  |  |

|                         |   | Area N               | ame and | l No.           | Α      | ircraft Serial No. | Date    |
|-------------------------|---|----------------------|---------|-----------------|--------|--------------------|---------|
| Phase                   |   |                      | TERY    | COMPARTMENT - 3 |        |                    |         |
| Inspect<br>Phase<br>No. | Inspection I  | Requirements         | Status  | Faults and/or R | emarks | Action Taken       | Initial |
| ALL                     | 1. Electrical wiring in nose compartment and behind instrument panel for chafing, deterioration of insulation and connector seals, and security of connections. |                      |         |                 |        |                    |         |
| ALL                     | 2. Electrical equipment shock mounts for deterioration, free throw, bottoming and security. Grounding straps or bands for damage and security of connections.   |                      |         |                 |        |                    |         |
| ALL                     | 3. Heat/defog ducts and security.   | and valve for damage |         |                 |        |                    |         |
| ALL                     | 4. Lines and hoses behind instrument panel for loose connections and chafing.   |                      |         |                 |        |                    |         |
|                         |   |                      |         |                 |        |                    |         |

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

| PH                       | ASE | E NO COCKPIT INTERIOR - 4               |  |        | id No.          | Aircraft Serial No. Date |              |  |         |
|--------------------------|-----|---|--|--------|-----------------|--------------------------|--------------|--|---------|
| Inspect<br>Phase<br>No's |     | Inspection                              | Requirements   | Status | Faults and/or F | Remarks                  | Action Taken |  | Inilial |
| ALL                      | 1.  |   | n mechanism functionally<br>s closed. Hinge pins for<br>nd distortion. |        |                 |                          |              |  |         |
| ALL                      | 2.  | Release cables security and adeq        | for chafing, damage,<br>uate lubrication.                              |        |                 |                          |              |  |         |
| ALL                      | 3.  | Door jettison han<br>copper safety wire | dles properly wired with e.  |        |                 |                          |              |  |         |
| ALL                      | 4.  | Seat adjustment positive movement tion. | mechanisms for wear,<br>nt, locking and lubrica-                       |        |                 |                          |              |  |         |
| 3,6                      | 5.  | Inspect seats<br>movement.              | for positive recline   |        |                 |                          |              |  |         |

| Phase                   | No       |  | Name an<br>OR 4 |               |         | Aircraft Serial No. | Dat                                   | te       |
|-------------------------|----------|--|-----------------|---------------|---------|---------------------|---------------------------------------|----------|
| Inspect<br>Phase<br>No. |          | Inspection Requirements  | Status          | Faults and/or | Remarks | Action Taken        |                                       | Initial  |
| 3,6                     | 6.       | Cockpit structure for damage, cracks an corrosion (plates, panels and doors opened for access). (Access panels 70, 71, and 76, Fig 1 | d<br>or         |               |         |                     |                                       |          |
|                         |          | access). (Access panels 70, 71, and 76, Fig 14).   | -               |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
| ALL                     | 7.       | Electrical wiring for chafing, deterioration an security (pedestal console).   | d               |               |         |                     |                                       |          |
|                         |          | security (pedestal console).   |                 |               |         |                     | 4                                     |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  | <u> </u>        |               |         |                     |                                       |          |
|                         | ļ        |  |                 |               |         | ,                   |                                       |          |
| ALL                     | 8.       | Check circuit breakers, switches and knobs for security and proper operation.  | )r              |               |         |                     |                                       | <u> </u> |
|                         | ŀ        |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     | · · · · · · · · · · · · · · · · · · · | ļ        |
|                         | <u> </u> |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
|                         |          |  | ļ               |               |         |                     |                                       |          |
|                         |          |  |                 |               |         |                     |                                       |          |
| 1                       |          |  |                 |               |         |                     |                                       |          |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

|                          |    | NO   | Area N<br>CABIN INTERIOR - 5   | ame ar | nd No.        | A       | rcraft Serial No. | Date | 2       |
|--------------------------|----|--|--|--------|---------------|---------|-------------------|------|---------|
| Inspect<br>Phase<br>No's |    | Inspection F   | Requirements   | Status | Faults and/or | Remarks | Action Taken      |      | Initial |
| 3,6                      | 1. | Cabin floor panels for cracks, dents, delamination and security. Check that aircraft has been defueled (TM 55-1520-210-23-1) prior to removal of floor panels over the fuel cells and inspect for corrosion. (All floor Access Panels, Fig 1-4.) |  |        |               |         |                   |      |         |
| 3,6                      | 2. | corrosion (sound)  | for damage, cracks and proofing removed and doors opened for accanels, Fig 1-4). |        |               |         |                   |      |         |
| 3,6                      | 3. | Cargo door windo<br>functionally check   | ow jettison mechanisms<br>ed.  |        |               |         |                   |      |         |
| ALL                      | 4. | Inspect cargo door worn or bent.   | r retainers for cracks,  |        |               |         |                   |      |         |
|                          |    |  |  |        |               |         |                   |      |         |

| PH                       | ASE NO  | UNDER FLOOR OF CO  | Name at |                 | Aircraft Serial No. |              | Date    |
|--------------------------|---|--|---------|-----------------|---------------------|--------------|---------|
| Inspect<br>Phase<br>No's | Inspec  | tion Requirements  | Status  | Faults and/or I | Remarks             | Action Taken | Initial |
| ALL                      | corrosion (flo  | Fuselage structure for damage, cracks and corrosion (floor panels removed for access). (Access Panels 72 and 73, Fig 1-4.)                           |         |                 |                     |              |         |
| ALL                      | 2. Area under floor for evidence of moisture accumulation. Drain holes for clogged condition. (Access Panels 72 and 73, Fig 1-4.) |  |         |                 |                     |              |         |
| 3,6<br>C                 | 3. Collective fric<br>Panel 76, Fig 1-  | tion liners for wear. (Access  |         |                 |                     |              |         |
| ALL<br>C                 | tubes, links,<br>arms, jacksha<br>corrosion, dai  | linkages, including pushpull<br>bellcranks, idlers, levers,<br>fts, force gradients, etc., for<br>nage and security. (Access<br>72 and 73, Fig 1-4.) |         |                 |                     |              |         |
| ALL<br>C                 | control linka   | hings and rod end in flight<br>ges for excessive play and<br>ess Panels 70, 71, 72 and 73,   |         |                 |                     |              |         |

| ŧ                        |    | NO                     | Area N<br>UNDER FLOOR OF CO                      |         |                 | A                                     | rcraft Serial No. | Date | •            |
|--------------------------|----|------------------------|--|---------|-----------------|---------------------------------------|-------------------|------|--------------|
| Inspect<br>Phase<br>No's |    | Inspection I           | Regulrements                                     | Status  | Faults and/or F | Remarks                               | Action Taken      |      | Initiat      |
| ATT                      | C  | 7                      | 61   |         |                 |                                       |                   |      |              |
| ALL<br>C                 | 6. |                        | kage for damage, wear, ation. (Access Panels 70, |         |                 |                                       |                   |      | <b></b>      |
|                          | i  | 71, 72 and 76, Fig 1-4 | £.)  |         |                 |                                       |                   |      |              |
|                          | 1  |                        |  |         |                 |                                       |                   |      |              |
| <u> </u>                 |    |                        |  |         |                 |                                       |                   |      |              |
| }                        | j  |                        |  |         |                 | *                                     |                   |      |              |
| ALL                      | 7. | Electrical wiring for  | or chafing, deterioration                        |         |                 | <del></del>                           |                   |      | <b></b>      |
| <b>!</b> .               |    |                        | ess Panels 71, 72 and 73,                        |         |                 |                                       |                   |      |              |
|                          | İ  | Fig 1-4.)              |  | <b></b> |                 | <del></del>                           |                   |      | <b> </b>     |
| }                        |    |                        |  |         |                 | · · · · · · · · · · · · · · · · · · · |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
| 3,6                      |    | ••                     |  |         |                 | *                                     |                   |      |              |
| 3,6                      | 8. | Panel 72, Fig 1-4.)    | urity and damage. (Access                        |         |                 |                                       |                   |      |              |
|                          |    | 1 unor (2, 1 ig 1-4.)  |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  |         |                 | ·                                     |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      |              |
|                          |    |                        |  | <b></b> |                 |                                       |                   |      | ļ            |
|                          |    |                        |  |         |                 | ~                                     |                   |      |              |
|                          |    |                        |  |         |                 |                                       |                   |      | <b> </b>     |
|                          |    |                        |  |         |                 | <del></del>                           |                   |      |              |
|                          |    |                        |  |         |                 | *                                     |                   |      | <b> </b>     |
|                          |    |                        |  |         |                 | M                                     |                   |      | <del> </del> |
|                          |    |                        |  |         |                 | <del>* - *</del>                      |                   |      | <b></b>      |

| PH                       | ASE | NO  | Area N<br>LOWER PYLON AREA   |        |                 | ^ | ircraft Serial No. | Date |         |
|--------------------------|-----|---|--|--------|-----------------|---|--------------------|------|---------|
| Inspect<br>Phase<br>No s |     | Inspection I  | Regulrements   | Status | Faults and/or F | l | Action Taken       |      | Initial |
| ALL                      | 1.  | Transmission movand deterioration. 24, Fig 1-4).  | int boots for cuts, tears<br>(Access Panels 3, 23 and                                    |        |                 |   |                    |      |         |
| 2,4,6<br>C               | 2.  | inspect resilient pylon mounts (5 each) for deterioration, cleanliness and security. (Access Panels 3, 23 and 24, Fig 1-4.) |  |        |                 |   |                    |      |         |
| ALL                      | 3.  | Friction dampers (security. (Access Par   | 2 each) for damage and<br>nels, 23 and 24, Fig 1-4.)                                     |        |                 |   |                    |      |         |
| ALL                      | 4.  | and fifth mount sup   | tural supports (4 places)<br>port fitting (1 each) visual-<br>rrosion. (Access Panels 23 |        |                 |   |                    |      |         |
| ALL                      | 5.  | Lift link for corros<br>ty. (Access Panel 2-  | sion, damage and securi-<br>4, Fig 1-4.)   |        |                 |   |                    |      |         |

|                          | ASE NO   | LOWER P<br>CABIN INT   | Name and<br>YLON A<br>(ERIOR) | REA (VIA      |         | Aircraft Serial No. | Date    |
|--------------------------|--|--|-------------------------------|---------------|---------|---------------------|---------|
| Imapeet<br>Phose<br>No.5 | lnspe  | ction Regulrements   | Status                        | Faults and/or | Remarks | Action Taken        | Initiat |
| 2,4,6<br>C               |  | k. Inspect bearings and attach<br>ir, cracks and tolerance. (Ac-<br>Fig 1-4.)        |                               |               |         |                     |         |
| ALL<br>C                 | 7. Lift beam vist<br>24 and 79, Fig            | nally for cracks. (Access Panel:<br>1-4 and Hell Hole.)                              |                               |               |         |                     |         |
| ALL<br>C                 | 8. Power turbine<br>wear and secu<br>Fig 1-4.) | governor controls for damage<br>rity. (Access Panels 23 and 24                       |                               |               |         |                     |         |
| 3,6<br>C                 | 9. Hydraulic fi<br>metal) replace              | lter element (either paper or<br>ed. (Access Panel 79, Fig 1-4.)                     |                               |               |         |                     |         |
| ALL                      | of insulatior                                  | ring for chafing, deterioration<br>and connector seals and<br>nnections (Hell Hole). |                               |               |         |                     |         |

|                         |   | Area M<br>LOWER PYLON AR<br>INTERIOR) — 7 (CC |        |               |                | Aircraft Serial No. |   | e             |
|-------------------------|---|---|--------|---------------|----------------|---------------------|---|---------------|
| Inspect<br>Phase<br>No. | Inspection R  | equirements                                   | Status | Faults and/or | Remarks        | Action Taken        |   | Initial       |
| 6                       | 11. Remove and replace<br>pin. (Access panel 24, Fig  | droop compensator shear                       |        |               |                |                     |   |               |
|                         | pin. (Access paner 24, Fig  | g 1 <del>-4)</del> .                          |        |               |                |                     |   |               |
|                         |   |   |        |               | <u></u>        |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
| ALL<br>C                | 12. All fuel supply lines   | for chafing, damage and                       |        |               | <del>~~~</del> |                     |   |               |
| C                       | 12. All fuel supply lines<br>leaks. Self-sealing li<br>swelling, blistered a<br>soaked with fuel or | areas that appear to be                       |        |               |                |                     |   | ļ             |
|                         | plies). (Access panel   | plies). (Access panel 23, Fig 1-4).           |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     | · | r             |
|                         |   |   |        |               |                |                     |   | ļ <del></del> |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   | -             |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               | <del></del>    |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        |               |                |                     |   |               |
|                         |   |   |        | :             |                |                     |   | <u> </u>      |
|                         |   |   |        |               |                |                     |   | -             |
|                         |   |   | 1      |               |                |                     |   | 1             |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

| Phase                   | e No UPPER PY  | Area Name and No.<br>LON (VIA CABIN R | OOF) — 8              | Aircraft Serial No. | Date    |
|-------------------------|--|---------------------------------------|-----------------------|---------------------|---------|
| Inspect<br>Phase<br>No. | inspection Requirements  | Status                                | Faults and/or Remarks | Action Taken        | Initial |
| ALL                     | Hydraulic reservoir, filler cap and s<br>condition. (Access panel 2, Fig 1-4).   | strainer for                          |                       |                     |         |
| ALL<br>C                | 2. Generator/alternator drive quill mag removed and visually checked for con Check vent on generator quill case fo (Access panels 2 and 3, Fig 1-4). | taminants.                            |                       |                     |         |
| ALL                     | 3. Generator/alternator electrical connections security. (Access panel 2, Fig 1-4).  | ections for                           |                       |                     |         |
| 1,3,5                   | 4. Generator.  |                                       |                       |                     |         |
| ALL                     | 5. Alternator.   |                                       |                       |                     |         |

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

| Phase                   |   | Area Name and No.  UPPER PYLON (VIA CABIN ROOF) — 8  (CONT) |                 |        | Aircraft Serial No. |  | 9       |
|-------------------------|---|---|-----------------|--------|---------------------|--|---------|
| inspect<br>Phase<br>No. | Inspection Requirements   | Status  | Faults and/or R | emarks | Action Taken        |  | Initial |
| ALL<br>C                | 6. Perform spring scale check on collective and cyclic hydraulic actuator support mount bearings P/N 204-076-168-1 (not required on cylinder assembly P/N 205-076-099).                         |   |                 |        |                     |  |         |
| ALL                     | <ol> <li>Transmission housings, fittings, and oil manifold<br/>for chafing damage and leaks. Check vent on<br/>top of transmission case for clogging. (Access<br/>panel 2, Fig 1-4).</li> </ol> |   |                 |        |                     |  |         |
| ALL<br>C                | 8. Visually inspect the Kamatics Main Drive Shaft.  |   |                 |        |                     |  |         |
| 6                       | 9. Deleted.   |   |                 |        |                     |  |         |
|                         |   |   |                 |        |                     |  |         |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

TM 55-1520-210-PM

| РН                       | PHASE NO       |                        | Area Name and No.  MAIN ROTOR AND MAST AREA - 9         |                |  | Aircraft Serial No. |              |   | 9            |
|--------------------------|----------------|------------------------|---|----------------|--|---------------------|--------------|---|--------------|
| Inspect<br>Phase<br>No's |                | Inspection F           | Requirements  | Status         | Faults and/or f                        | l<br>Remarks        | Action Taken | L | Initial      |
| ALL                      | 1.             | Break torque on        | stabilizer bar support                                  |                |  |                     |              |   |              |
| С                        |                | mount bolts (8         | each) and torque to                                     |                |  |                     |              |   |              |
| 1                        | [              | specifications.        |   | <del> </del>   |  |                     |              |   |              |
|                          | •              |                        |   | -              |  |                     |              |   | <del> </del> |
|                          |                |                        |   |                |  |                     |              |   |              |
| ALL                      | 2.             | Check for excessiv     | e play in trunnion bear-                                | $\vdash$       | <del></del>                            |                     |              |   |              |
| С                        | 1              | ings, collective lev   | er bearings, and for ex-                                |                |  |                     |              |   |              |
|                          | ł              | drive plate and mas    | ween collective sleeve st.                              |                |  |                     |              |   |              |
|                          | ĺ              |                        |   | <b>  </b>      | ************************************** |                     |              |   |              |
|                          | <del> </del> - |                        |   |                | ورنواده الانتقاف المسهودة المساوي      |                     |              |   |              |
| 47.7                     | ,              | Calaria 1 at           | ve assembly for visible curity. Bearings and sive play. | <del>  -</del> |  |                     |              |   |              |
| ALL<br>C                 | 1              |                        |   |                |  |                     |              |   |              |
|                          |                |                        |   |                |  |                     |              |   |              |
|                          |                |                        |   |                |  |                     |              |   |              |
| <b> </b>                 |                |                        |   |                |  |                     |              |   |              |
| ALL                      | 4.             | Disconnect scissor     | rs drive links from trun-                               | <b></b>        |  |                     |              |   |              |
| C                        |                | nions and check s      | swashplate bearing for                                  |                |  |                     |              |   |              |
|                          |                | roughness, binding     | g and vertical play.                                    |                |  |                     |              |   |              |
| 1                        |                |                        |   |                |  |                     |              |   |              |
|                          |                |                        |   |                |  |                     |              |   |              |
|                          |                |                        |   |                |  |                     |              |   |              |
| ALL                      | 5.             |                        | ly for security, condition,                             |                |  |                     |              |   |              |
| C                        | defo           | ormation, and cracks i | n rubber bumbers.                                       |                |  |                     |              |   |              |
|                          |                |                        |   |                |  |                     |              |   |              |
|                          |                |                        |   |                | ************************************** |                     |              |   |              |
| <b></b>                  | <u> </u>       |                        |   | L              |  |                     |              |   |              |

| Phase No MAIN ROTOR AND N |    |   |          |                 | Aircraft Serial No. |              | Date |         |
|---------------------------|----|---|----------|-----------------|---------------------|--------------|------|---------|
| Inspect<br>Phase<br>No.   |    | Inspection Requirements   |          | Faults and/or I | Remarks             | Action Taken |      | Initial |
| ALL                       | 6. | Visually inspect composite main rotor blades  |          |                 |                     |              |      |         |
| C                         |    | for evidence of debonding of the leading edge abrasion strip, trim tab, and taco patch. Inspect   |          |                 |                     |              |      |         |
|                           |    | Visually inspect composite main rotor blades for evidence of debonding of the leading edge abrasion strip, trim tab, and taco patch. Inspect tie down plate and aft root weights for security. Inspect leading and trailing edge for dents and nicks. |          |                 |                     |              |      |         |
|                           |    | dents and nicks.  |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           | 7. | Inspect metal main rotor blades.  |          |                 |                     |              |      |         |
| C                         |    |   | <u> </u> |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              | –    |         |
|                           |    |   |          |                 |                     |              |      |         |
| 4                         | 8. | Remove tip cap, check stud retention nuts for looseness. Check studs for looseness or   |          |                 |                     |              |      |         |
|                           |    | looseness. Check studs for looseness or distortion.   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
| ļ                         |    |   |          |                 |                     |              |      |         |
| 1                         |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     | ,            |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |
|                           |    |   |          |                 |                     |              |      |         |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

| PH                       | ASE NO   |   | Area N<br>ENGINE AIR INDUCT  |        | ame and No. Aircraft Serial No. ON AREA - 10 |         | ircraft Serial No. | Date    |         |
|--------------------------|--|---|--|--------|--|---------|--------------------|---------|---------|
| Inspect<br>Phase<br>No's |  | Inspection  | Requirements   | Status | Faults and/or F                              | Remarks | Action Taken       | <b></b> | Initial |
| ALL<br>C                 | Air induction baffle assembly for chafing, cracks, dents loose or missing fasteners and security. (Access Panel 4, Fig 1-4.) |   |  |        |  |         |                    |         |         |
| ALL<br>C                 | 2.   | 2. Particle separator disassembled and inspected for clogging and damage. Gaskets and seals for cuts deterioration and separation from backing plates. (Access Panel 4, Fig 1-4). |  |        |  |         |                    |         |         |
| ALL<br>C                 | 3.   | separator) remove   | self-purging particle<br>d, cleaned and inspected<br>ss Panel 4, Fig 1)4.) |        |  |         |                    |         |         |
| 3,6                      | 4.   | of insulation an  | for chafing deterioration d connector seals, and ections. (Access Panel 4, |        |  |         |                    |         |         |
|                          |  |   |  |        |  |         |                    |         |         |

| PH                       | Area NOENGINE AIR INDUCT              |  |   |          |                 | Aircraft Serial No. |              | 2            |              |
|--------------------------|---------------------------------------|--|---|----------|-----------------|---------------------|--------------|--------------|--------------|
| Inspect<br>Phase<br>No a |                                       | Inspection F   | Requirements                                    | Status   | Faults and/or F | Remarks             | Action Taken |              | Initial      |
|                          |                                       |  |   |          |                 |                     |              |              |              |
| l                        | ļ                                     | CAU  | TION  |          |                 |                     |              |              |              |
| l                        | ł                                     | (T) e .1   |   |          |                 |                     |              |              | <del> </del> |
| ŀ                        |                                       |  | next inspection re-<br>igine variable inlet     |          |                 |                     |              |              |              |
|                          | l                                     |  | be positioned to the                            |          |                 |                     |              |              |              |
|                          |                                       | full open positi   | ion. To preclude                                |          |                 |                     |              |              |              |
| 1                        |                                       | damage/distortion to VIGV com-<br>ponents, release torques on "B" nuts       |   |          |                 |                     |              |              |              |
|                          | to CYL 1 and CYL 2 lines at actuator. |  |   |          |                 |                     |              |              |              |
|                          |                                       |  | <b></b>   |          |                 |                     |              | <b> </b> -   |              |
|                          |                                       |  |   |          |                 |                     |              | <del> </del> |              |
|                          |                                       |  |   | <b></b>  |                 |                     |              |              |              |
| ALL<br>C                 | 5.                                    |  | housing, variable inlet                         |          |                 | <u> </u>            |              |              | <del> </del> |
|                          |                                       | guide vanes and first stage compre<br>blades for foreign object damage, eros |   |          |                 |                     |              |              | <b></b> -    |
| ] ]                      |                                       |  | its and oil streaks. (Ac-                       |          |                 |                     |              |              |              |
|                          |                                       | cess Panel 4, Fig 1-   |   |          | ·               | <del></del>         |              |              |              |
| <u> </u>                 |                                       |  |   |          |                 |                     |              | 9            |              |
|                          |                                       | //   | .:L1. d   |          |                 |                     |              |              |              |
| ALL<br>C                 | 6.                                    |  | sible damage, cleanness, curity of seals around | <b> </b> |                 | ·                   |              |              |              |
|                          |                                       | edges. (Access Pan   |   |          |                 |                     |              |              | <del> </del> |
| j 1                      |                                       | <b>J</b> , , , , , , , , , , , , , , , , , , ,                               | . <b>.</b> .                                    | <b></b>  |                 | <del></del>         |              |              | <del> </del> |
|                          |                                       |  |   |          |                 |                     |              |              | <del> </del> |
|                          |                                       | <del></del>  |   |          |                 |                     |              |              |              |
| ALL                      | 7.                                    | Right air filter fo  | r visible damage clean-                         |          |                 |                     |              | ·            | <del> </del> |
| C                        | •••                                   |  | and security of seals                           |          |                 |                     |              |              | <b></b>      |
| Ĭ                        |                                       |  | ess Panel 4, Fig 1-4.)                          |          |                 |                     |              |              |              |
|                          |                                       |  |   | <b></b>  |                 |                     |              |              |              |
| LI                       | L                                     |  |   |          |                 |                     |              |              |              |

| 1                       | e No                           | Area N<br>ENGINE AIR INDUC   | lame an |                 | A      | ircraft Serial No. | Date    |
|-------------------------|--------------------------------|--|---------|-----------------|--------|--------------------|---------|
| Inspect<br>Phase<br>No. | Inspection I                   | Requirements   | Status  | Faults and/or R | emarks | Action Taken       | Initial |
| ALL<br>C                | 8. Left air filter for v       | 8. Left air filter for visible damage, cleanness, condition and security of seals around edges (Access Panel 4, Fig. 1-4). |         |                 |        |                    |         |
| ALL<br>C                | 9. IMPROVED PARTICLE SEPARATOR |  |         |                 |        |                    |         |
|                         |                                |  |         |                 |        |                    |         |
|                         |                                |  |         |                 |        |                    |         |

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

| Phase                   | No | )   | Area N<br>ENGINE COMPARTI                     | ame and No.<br>MENT ] |               | Air     | craft Serial No. | Date   |
|-------------------------|----|---|---|-----------------------|---------------|---------|------------------|--------|
| Inspect<br>Phase<br>No. |    | Inspection Re   | quirements                                    | Status                | Faults and/or | Remarks | Action Taken     | Initia |
| ALL<br>C                |    | <ol> <li>Engine airbleed actuator strainer for condition and cleanliness. Bleed band assembly for bends, cracks and security. (Access panels 6 and 7, Fig 1-4).</li> <li>Starter-generator.</li> <li>Fuel control inlet strainer inspect and clean. (Access panel 7, Fig 1-4).</li> </ol> |   |                       | _             |         |                  |        |
| 1,3,5                   | 2. |   |   |                       |               |         |                  |        |
| ALL<br>C                | 3. |   |   |                       |               |         |                  |        |
| ALL<br>C                | 4. | Inspect and clean fuel of replace filter. (Access p   | control servo strainer and panel 7, fig 1-4). |                       |               |         |                  |        |
|                         | 5. | Deleted   |   |                       |               |         |                  |        |

"FOD REMINDER" Check work area for tools and part after completion of maintenance and inspection.

|          | ASI | E NO   | Area N<br>ENGINE COMPARTMI  | lame and Mo.<br>ENT-11 (CON | T                     | Aircraft Serial No. | Date   |
|----------|-----|--|---|-----------------------------|-----------------------|---------------------|--------|
| 33       |     | (nepretium (   |   | Status                      | Faniha andler Remarks | Action Tobaca       | termer |
| ALL<br>C | 6.  | leaks and securi   | and hoses for chafing,<br>ity. Braided hoses for<br>wires. (Access Panesl 6   |                             |                       |                     |        |
| ALL<br>C | 7.  | Main fuel filter mis<br>spected and replace<br>type. (Access Panel | cronic paper element in-<br>ed. Clean if metal screen<br>7, Fig 1-4.)   |                             |                       |                     |        |
| ALL<br>C | 8.  | valve (1 each) outle<br>(crashworthy only)<br>and cracks in brea   | and breakaway type  t of main fuel strainer for leakage, security akable (necked) section vay pins in outer sleeve Panel 7, Fig 1-4.) |                             |                       |                     |        |
| ALL<br>C | 9.  | main fuel strainer<br>security, leakage as<br>(necked) section as  | elve (1 each) on inlet of<br>(crashworthy only) for<br>and cracks in breakable<br>and play in breakaway<br>e staked area. (Access     |                             |                       |                     |        |
| ALL<br>C | 10. | oil tank inlet from<br>(crashworthy only)<br>and cracks in brea    | alve (1 each) on engine<br>engine breather hose<br>for security, leakage<br>akable (necked) section<br>way pins in outer sleeve       |                             |                       |                     |        |

| Phase                   | No.      | Area P<br>ENGINE COMPART   | Name and | 7               |                                       | Aircraft Serial No. | Dat | te           |
|-------------------------|----------|--|----------|-----------------|---------------------------------------|---------------------|-----|--------------|
| Inepect<br>Phase<br>No. |          | Inspection Requirements  | Status   | Faults and/or f | Remarks                               | Action Taken        |     | Initial      |
| ALL                     | 10.A     | . Self-sealing oil system component hoses for  |          |                 |                                       |                     |     |              |
| С                       | appe     | activation (e.g., swelling, blistering, areas that ar to be soaked with oil).  |          |                 |                                       |                     |     |              |
|                         | !        |  |          |                 |                                       |                     |     |              |
|                         |          |  |          |                 | · · · · · · · · · · · · · · · · · · · |                     |     |              |
| ALL<br>C                | 11.      | Engine Oil Filter Inspect (TM 55-2840-229-23).   |          |                 |                                       |                     |     |              |
|                         |          | 23).   |          |                 |                                       |                     |     |              |
|                         |          |  |          |                 |                                       |                     |     |              |
| ALL                     | 12.      | Production to the product of the pro |          | -               |                                       |                     |     |              |
| C                       | 12.      | scavenge pump outlet hose and thermal bypass<br>input line (crashworthy only) at engine deck<br>for security, leakage, and cracks in breakable<br>(necked) section and play in breakaway pins in<br>outer sleeve area. (Access panels 7 and 53, Fig  |          |                 |                                       |                     |     |              |
|                         |          |  |          |                 |                                       |                     |     | -            |
|                         | 1-4).    |  |          |                 |                                       |                     |     |              |
|                         | 1-7).    |  |          |                 |                                       |                     |     |              |
| ALL<br>C                | 13.      | Breakaway type valve (1 each) on engine oil tank outlet to engine (crashworthy only) for   |          |                 |                                       |                     |     |              |
|                         |          | leakage, security, and cracks in breakable (necked) section and play in breakaway pins in  | ]        |                 |                                       |                     |     |              |
|                         |          | outer sleeve staked area.  |          |                 |                                       |                     |     | <del> </del> |
|                         |          |  | <u> </u> |                 |                                       |                     |     |              |
|                         |          |  |          |                 |                                       |                     |     |              |
| 2,4,6                   | 14.      | <ol> <li>Engine mount rod ends for maximum<br/>allowable axial and radial play (Access Panel<br/>7, Fig 1-4).</li> </ol>   |          |                 | <u> </u>                              |                     |     |              |
|                         | <u>;</u> |  |          |                 |                                       |                     |     |              |
|                         |          |  |          |                 | ····                                  |                     |     |              |
|                         |          |  |          |                 |                                       |                     |     |              |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

| PHASE                    | NO   | ENGINE C   | Area Name and No. COMPARTMENT -11 (Cont) | Aircraft Serial No. | Date    |
|--------------------------|--|--|--|---------------------|---------|
| Inspect<br>Phase<br>No's | Inspection Requirements  | Inspection Requirements Status Faults and/or Remarks |  | Action Taken        | Initial |
| 2,4,6,                   | 15. Engine mount pillow block assemblies for wear and damage. Trunnion caps for damage and security. Trunnion bearings for wear and excessive axial and radial play (Access Panels 6 and 7, Fig. 1-4). |  |  |                     |         |
| 2,4,6,                   | 16. Engine work platform decks for bonding separation, cracks, punctures and corrosion (Access Panels 7 and 10, 19 and 22, Fig. 1-4).  |  |  |                     |         |
| ALL                      | 17. Inspect mono, biped, and tripod engine deck mounting pads and attaching hardware for; looseness and security. If looseness is evident, check bolts and holes for damage.                           |  |  |                     |         |
| ALL                      | 18. Engine deck drain holes and channels for obstruction.  |  |  |                     |         |
| ALL                      | 18.1 Inspect Break-Away Coupling<br>(90-degree and straight halves) at the<br>ODDS Lubriclone Filter. Check wear<br>on the Break-Away Pins.  |  |  |                     |         |

| Phase                   | e No                                       | Area N<br>ENGINE COMPART                              | lame an | d No.<br>' — 11 (CONT) |         | Aircraft Serial No. | Dat                                   | te       |
|-------------------------|--|---|---------|------------------------|---------|---------------------|---------------------------------------|----------|
| Inspect<br>Phase<br>No. | Inspection F                               | Requirements  | Status  | Faults and/or i        | Remarks | Action Taken        |                                       | initial  |
| ALL                     | 19. Power turbine gover                    | rnor control tube, levers, points for wear, security, |         |                        |         |                     |                                       |          |
|                         | and corrosion.                             | points for wear, security,                            |         |                        |         |                     |                                       |          |
|                         | ĺ  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       | <u> </u> |
|                         | 20 Droop compensator for proper attachment |   |         |                        |         |                     | <del></del>                           |          |
| ALL                     | 20. Droop compensator lube, and corrosion. | for proper attachment,                                |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       |          |
| ALL                     | 21. Linear actuator i                      | for security, electrical d proper operation.          |         |                        |         |                     | <del></del>                           |          |
| Ì                       | connections, wear an                       | d proper operation.                                   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       |          |
| }                       |  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     | <del></del>                           |          |
|                         |  |   |         |                        |         |                     | <del></del>                           |          |
|                         |  |   |         |                        |         |                     | ·                                     |          |
|                         |  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     |                                       |          |
|                         |  |   |         |                        |         |                     | · · · · · · · · · · · · · · · · · · · |          |
|                         |  |   |         |                        |         |                     | <del></del>                           |          |
|                         |  |   |         |                        |         |                     |                                       |          |

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

| Phase                   | No                | Area<br>CABIN ROOF — 12 | Name an  | d No.           |         | Aircraft Serial No. | Date        |         |
|-------------------------|-------------------|-------------------------|----------|-----------------|---------|---------------------|-------------|---------|
| Inspect<br>Phase<br>No. | Inspection R      | lequirements            | Status   | Faults and/or I | Remarks | Action Taken        |             | Initial |
|                         | NO                | )TE                     |          |                 |         |                     |             |         |
|                         | This page intenti | onally left blank.      |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         | <u> </u> |                 |         |                     |             |         |
|                         |                   |                         | ļ        |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             | ļ       |
| ļ                       |                   |                         |          | <u> </u>        |         |                     |             |         |
|                         |                   |                         | -        |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             | ,       |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         | -                   | <del></del> | ļ       |
|                         |                   |                         |          |                 |         |                     |             | ļ       |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         | <b> </b> |                 |         |                     |             |         |
|                         |                   |                         | -        |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             | <u></u> |
|                         |                   |                         |          |                 |         |                     |             |         |
|                         |                   |                         |          |                 |         |                     |             |         |

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

| PHASE                    | NO  | CABIN E<br>GEAR - | Area Name and No.<br>XT SIDES, BOTTOM, & LANDING<br>13 (CONT) | Aircraft Serial No. | Da | ite     |
|--------------------------|---|-------------------|---|---------------------|----|---------|
| Inspect<br>Phase<br>No's | Inspection Requirements   | Status            | Faults and/or Remarks   | Action Taken        |    | Initial |
| 3,6                      | 5. Landing gear crosstubes for exxcessive spread. (Check measurements).   |                   |   |                     |    |         |
| ALL<br>C                 | 6. Fuel boost pumps (electrical and air drive) visually for leaks, damage and security. (access panel 81, Fig 1-4)  |                   |   |                     |    |         |
| ALL<br>C                 | 7. Retention cap assemblies and bearing Support fittings at crosstube to fuselage, attach points for deterioration, wear an d security.                                       |                   |   |                     |    |         |
| 3,6                      | 8. External stores support assembly beams when installed, for fatigue cracks within 6 inches of upper fuselage attach points. Use non-destructive (TM 55-1500-335-23 method). |                   |   |                     |    |         |
| ALL<br>C                 | Landing gear skid shoes for wear, damage and security.  |                   |   |                     |    |         |
| ALL<br>C                 | 10. Landing gear skid tubes (skid shoes removed), saddles, steps, tow rings, end caps and ground handling wheel attach lugs for cracks, damage and security.                  |                   |   |                     |    |         |

2-36

| PHASE                    | NO  | UNDE   | Area Name and No.<br>R CABIN PYLON (HELL HOLE)-14 | Aircraft Serial No. | Da | ate     |
|--------------------------|---|--------|---|---------------------|----|---------|
| Inspect<br>Phase<br>No's | Inspection Requirements   | Status | Faults and/or Remarks                             | Action Taken        |    | Initial |
| ALL<br>C                 | 1. Fuel system components and associated lines and hoses for chaffing, damage, leaks and security. (Access panels 48, 50, 52, 57 and 59, Fig. 1-4.)   |        |   |                     |    |         |
| ALL<br>C                 | 2. Self-sealing fuel system component lines and hoses for activation (e.g., swelling, blistering, areas that appear to be soaked with fuel or have fuel between fabric plies). (Hell Hole)          |        |   |                     |    |         |
| ALL<br>C                 | 3. Fuel supply and particle separator discharge lines and highest protruding attaching hardware in the area of transmission sump and tail rotor shaft for 0.5 inch clearance with sump. (Hell Hole) |        |   |                     |    |         |
| ALL<br>C                 | 4. Transmission oil lines and hoses for chafing, damage and leaks. Transmission oil line quick disconnects for excessive play.  |        |   |                     |    |         |
| ALL<br>C                 | 5. Transmission lower housing and fittings for chafing, damage and leaks. (Hell Hole)   |        |   |                     |    |         |

| Phase                   | No.        |  |  | Name and No.<br>LON (HELL HOLE) 14 |   |                                       | Aircraft Serial No. | Date        |          |
|-------------------------|------------|--|--|------------------------------------|---|---------------------------------------|---------------------|-------------|----------|
| inepect<br>Phase<br>No. |            | Inspection Requirements  |  |                                    | Faults and/or R                         | emarks                                | Action Taken        |             | Initial  |
|                         | 6.         | Deleted  |  |                                    |   |                                       |                     |             |          |
| •                       | ]          |  |  |                                    |   | · · · · · · · · · · · · · · · · · · · |                     |             |          |
| ALL                     | 7.         | Transmission extern  | nal oil filter element                           |                                    |   |                                       |                     |             |          |
| С                       | ŀ          | replaced, (Hell Hole). (Aircraft Withou<br>ODDS).                        |  |                                    | ·                                       |                                       |                     |             |          |
| ALL                     | 7.1        | 7.1 Inspect transmission debris monitor, aircraft with ODDS.             |  | -                                  |   |                                       |                     |             | ļ . ———— |
| C                       | <b>'''</b> |  |  |                                    |   |                                       |                     |             |          |
| •                       |            |  |  |                                    |   |                                       |                     |             |          |
|                         |            |  |  |                                    |   |                                       |                     |             |          |
| ALL<br>C                | 8.         | Bearings, bushings<br>control linkages<br>security. (Hell Hole)          | and rod ends in flight<br>for excessive play and |                                    |   |                                       |                     |             |          |
|                         |            | security. (Hell Hole)  |  |                                    |   |                                       |                     |             |          |
|                         | <u> </u>   | · · · · · · · · · · · · · · · · · · ·                                    | ····   |                                    |   |                                       |                     |             |          |
| ALL<br>C                | 9.         | Flight control links   | ages including push-pull                         |                                    | , |                                       |                     |             |          |
|                         |            | assemblies, etc., for security. (Hell Hole)                              | ks, idlers, support corrosion, damage and        |                                    |   |                                       |                     |             | <u> </u> |
|                         |            | security. (near note).   |  |                                    |   |                                       |                     |             |          |
| 246                     | 10         | 10. Throttle control linkage for damage, wear and security. (Hell Hole). |  |                                    |   |                                       |                     | <del></del> |          |
| 2.4.6<br>C              | 10.        |  |  |                                    |   |                                       |                     |             |          |
|                         |            |  |  |                                    |   |                                       |                     |             |          |
|                         |            |  |  |                                    |   |                                       |                     | ******      |          |

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

| i                        |          | NO                  | Area N<br>UNDER CABIN PYLON              | ame an<br>(HEL |               | A                                      | ircraft Serial No. | Date        | P            |
|--------------------------|----------|---------------------|--|----------------|---------------|--|--------------------|-------------|--------------|
| Inspect<br>Phase<br>No a |          | Inspection I        | Requirements                             | Status         | Faults and/or | Remarks                                | Action Taken       |             | Initial      |
| 2,4,6                    | 11.      | Electrical wiring f | or chafing deterioration                 |                |               |  |                    |             |              |
|                          |          | security of connect | l connector seals, and ions. (Hell Hole) | }              |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    | <del></del> |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  | ,              |               |  |                    |             |              |
| 2,4,6                    | 12.      | Cargo suspension    | assembly for damage                      |                |               |  |                    |             |              |
| С                        |          | and security. (Hell | Hole)                                    |                |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  |                |               | ······································ |                    |             |              |
|                          |          |                     |  |                |               | <del> </del>                           |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             | ļ            |
|                          |          |                     |  |                |               |  |                    | <del></del> | <del> </del> |
|                          |          |                     |  |                |               |  |                    |             | <del> </del> |
|                          |          |                     |  | <b></b>        |               | ···                                    |                    | <del></del> |              |
|                          |          |                     |  |                |               |  |                    |             | <del> </del> |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          | <u> </u> |                     |  | ļ              |               |  |                    |             |              |
|                          |          |                     |  |                |               |  |                    |             |              |
|                          |          |                     |  | <del> </del>   |               |  |                    |             |              |
|                          |          |                     |  |                | <del></del>   | <del></del>                            |                    |             |              |
| 1                        |          |                     |  |                |               |  |                    |             |              |
|                          | L        |                     |  |                |               |  |                    |             |              |

| PH                       | PHASE NO |  | Area Name and No. MID-FUSELAGE UNDER ENG DECK - 15 |        |               | , A     | Aircraft Serial No. |  | e       |
|--------------------------|----------|--|--|--------|---------------|---------|---------------------|--|---------|
| Inspect<br>Phase<br>No's |          | Inspection   | Requirements                                       | Status | Faults and/or | Remarks | Action Taken        |  | Initial |
| ALL                      | 1.       | Fuselage structure behind cabin and below engine deck for damage, cracks and corrosion. (Access panel 53, Fig. 1-4).                     |  |        |               |         |                     |  |         |
| ALL<br>C                 | 2.       | Throttle control linkage for damage, wear and security. (Access panel 53, Fig. 1-4).   |  |        |               |         |                     |  |         |
| ALL                      | 3.       | Electrical wiring for chafing, deterioration of insulation and connector seals, and security of connections. (Access panel 53, Fig.1-4). |  |        |               |         |                     |  |         |
| 2,4,6<br>C               | 4.       | Engine idle solenoid for operation, freedom of plunger, corrosion and security. Check for proper clearance.                              |  |        |               |         |                     |  |         |
| ALL<br>C                 | 5.       | Bleed air lines for condition and security. (Access panel 53, Fig. 1-4).   |  |        |               |         |                     |  |         |

|                          | ASE NO  | Area N<br>MID-FUSELAGE UNDI | ame and No | ).<br>CK - 15 (CONT) | Aircra | ilt Serial No. | Date | P       |
|--------------------------|---|-----------------------------|------------|----------------------|--------|----------------|------|---------|
| Inspect<br>Phase<br>No's | Inspection  | n Requirements              | Status     | Faults and/or Re     | marks  | Action Taken   | ·L   | Inilial |
|                          | Deleted.  |                             |            |                      |        |                |      |         |
| ALL<br>C                 | 7. Fuel system lines and hoses for chafing, leaks and security. Braided hoses for frayed or broken wires. (Access panel 53, Fig.1-4). |                             |            |                      |        |                |      |         |
|                          |   |                             |            |                      |        |                |      |         |

| PH.                      | ASE  | NO  | Area N<br>ELECTRONIC/COMM  | ame ar  | od No.<br>PARTMENT - 16               | A           | Aircraft Serial No. |  | 2           |
|--------------------------|------|---|----------------------------|---------|---------------------------------------|-------------|---------------------|--|-------------|
| Inspect<br>Phase<br>No's |      | Inspection F  | lequirements               | Status  | Faults and/or Remarks                 |             | Action Taken        |  | Initial     |
| 3,6                      | 1.   | Fuselage structure for damage, cracks and corrosion. (Access panels 19, 21 and 22, Fig. |                            |         |                                       |             |                     |  |             |
|                          |      | 1-4).   | paneis 19, 21 and 22, rig. | <b></b> |                                       |             |                     |  |             |
|                          |      |   |                            |         | · · · · · · · · · · · · · · · · · · · |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  | <b></b>     |
|                          |      |   |                            |         |                                       | ···         |                     |  |             |
|                          |      | Deleted.  |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  | <b> </b> -  |
|                          |      |   |                            |         |                                       |             |                     |  | <del></del> |
|                          |      |   |                            |         |                                       |             |                     |  |             |
| 1                        |      |   |                            |         |                                       |             |                     |  |             |
| į                        |      |   |                            |         |                                       | <del></del> |                     |  | <del></del> |
|                          |      |   |                            |         |                                       |             |                     | ······································ | <b>!</b>    |
| - 1                      |      |   |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     | <del></del>                            |             |
|                          |      |   |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  |             |
| I                        |      |   |                            |         |                                       |             |                     |  |             |
| l                        |      |   |                            |         |                                       | ····        |                     |  |             |
| 1                        |      |   |                            |         |                                       |             |                     |  |             |
|                          | **** |   |                            |         |                                       |             |                     |  |             |
|                          |      |   |                            |         |                                       |             |                     |  |             |
| l                        |      |   |                            |         |                                       | ·           |                     |  |             |
| 1                        |      |   |                            | <b></b> |                                       |             |                     |  |             |
| [                        |      |   | •                          |         |                                       |             |                     |  |             |
| I                        |      |   |                            |         |                                       |             |                     |  |             |
| l                        |      |   |                            | L       |                                       |             |                     |  |             |

|                          |   | E NO   | Area N<br>ENGINE AREA EXTEI | lame ar<br>RIOR - |               | A                                      | ircraft Serial No. | Date           | 2           |
|--------------------------|---|--|-----------------------------|-------------------|---------------|--|--------------------|----------------|-------------|
| Inspect<br>Phase<br>No's |   | Inspection I   | Requirements                | Status            | Faults and/or | Remarks                                | Action Taken       |                | leilini     |
| ALL                      |   |  |                             |                   |               |  |                    |                |             |
| •                        |   | and burned or buckled areas. (Access panel 8, Fig. 1-4). |                             | <b> </b>          |               |  |                    |                |             |
|                          |   | 0,1 1g.1 -4/.  |                             | <b> </b>          |               |  |                    |                |             |
|                          |   |  |                             | ļ                 |               |  |                    |                | <del></del> |
|                          |   |  | -                           |                   |               |  |                    | <del> </del> - |             |
| ALL                      | 2   | Uset exchange for  |                             |                   |               |  |                    |                |             |
| ALL                      | ٤.  | and inspect for cra                                      | r muff heater, remove       |                   |               |  |                    |                |             |
|                          | and inspect for cracks. (Access panels 8 and 10, Figure 1-4). |  |                             |                   | <del></del>   |  |                    |                |             |
|                          | and 10, Figure 1-4).  |  |                             |                   |               |  |                    |                |             |
| 1                        |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             | <u> </u>          |               |  |                    |                |             |
|                          |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             | <b></b>           |               |  |                    |                |             |
|                          |   |  |                             | ļ                 |               |  |                    |                |             |
|                          |   |  |                             | <b></b>           |               | ······································ |                    |                |             |
|                          |   |  |                             |                   |               | <del></del>                            |                    |                | <b> </b>    |
|                          |   |  |                             |                   |               | ···                                    |                    |                |             |
|                          |   |  |                             | ļ                 |               |  |                    |                |             |
|                          |   |  |                             | <b> </b>          |               |  |                    |                |             |
| 1                        |   |  |                             | <b></b>           |               |  |                    |                |             |
|                          |   |  |                             |                   |               | <del></del>                            |                    |                | <b></b>     |
|                          |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             | <del> </del>      |               |  |                    |                |             |
|                          |   |  |                             |                   |               | <del></del>                            |                    |                |             |
|                          |   |  |                             |                   |               |  |                    |                |             |
|                          |   |  |                             |                   |               | <del></del>                            |                    |                |             |
|                          |   |  |                             |                   |               |  | L                  |                |             |

| Indian   I   |                          | PHASE NO                                |  | Area Name and No. TAILBOOM INTERIOR - 18 |              | A               | ircraft Serial No.                    | Date               | e            |          |
|--|--------------------------|---|--|--|--------------|-----------------|---------------------------------------|--------------------|--------------|----------|
| C for corrosion, cracks and damage. (Access panels 16; 17, 18 and 36, Fig.1-4).  ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C Synchronized elevator control linkage for damage, binding, corrosion, and loose, missing or improperly installed hardware.  | Inspect<br>Phase<br>No's |   | Inspection I   | Requirements                             | Status       | Faults and/or F | lemarks                               | narks Action Taken |              | Initial  |
| ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4).   | ALL                      | 1.                                      | Tailboom structu   | re, including longerons                  |              |                 |                                       |                    |              |          |
| ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C Synchronized elevator control linkage for damage, binding, corrosion, and loose, missing or improperly installed hardware.   | С                        | [                                       | for corrosion, cracks and damage. (Access  |  |              |                 |                                       |                    |              | <u> </u> |
| C for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | ł                        |   | panels 16, 17, 18 and 36, Fig. 1-4).   |  | <b> </b> -   |                 |                                       |                    |              |          |
| C for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | 1                        | l                                       |  |  | <b></b>      |                 |                                       |                    |              |          |
| C for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | 1                        |   |  |  | <b> </b>     |                 |                                       |                    | <del></del>  | ļ        |
| C for corrosion and damage. (Access panel 17, Fig. 1-4).  ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | ΔΙΙ                      | 2                                       | Synchronized elevator supports (4 each)  |  | <del> </del> |                 | <del>,,,,,,,, </del>                  |                    |              |          |
| ALL Control of the state of the | •                        | <b> </b>                                | 2. Synchronized elevator supports (4 each) for corrosion and damage. (Access panel |  |              |                 | <del></del>                           |                    | ·            |          |
| C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | 1                        | for corrosion and damage. (Access panel |  | <b> </b>                                 |              | <del></del>     |                                       | <del></del>        |              |          |
| C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | }                        | 17, Fig. 1949.                          |  |  |              |                 |                                       |                    |              |          |
| C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | 1                        |   |  |  |              |                 |                                       |                    | <del> </del> |          |
| C damage, binding, corrosion, and loose, missing or improperly installed hardware.   | <b></b>                  |   |  |  |              |                 |                                       |                    |              |          |
| missing or improperly installed hardware.  |                          | 3.                                      | damage, binding, corrosion, and loose, missing or improperly installed hardware.   |  | L            |                 |                                       |                    |              | <u> </u> |
| (Access panels 17, 18 and 36, Fig.1-4).  | C                        | ł                                       |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  | ļ            |                 |                                       |                    |              |          |
|  |                          |   |  |  | <b></b>      |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 | <del></del>                           |                    |              | ļ        |
|  |                          |   |  |  |              |                 |                                       |                    |              | <u> </u> |
|  | i                        |   |  |  |              |                 | ·                                     |                    |              |          |
|  |                          |   |  |  |              |                 | · · · · · · · · · · · · · · · · · · · |                    |              | <b> </b> |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  | <b>]</b>                 |   | ···  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |
|  |                          |   |  |  |              |                 |                                       |                    |              |          |

| РН                       | PHASE NO TAILBOOM INTERIO |  | Area N<br>TAILBOOM INTERIO | lame and<br>R - 18 (CO |                 |        | ircraft Serial No. | Date | e       |
|--------------------------|---------------------------|--|----------------------------|------------------------|-----------------|--------|--------------------|------|---------|
| Inspect<br>Phase<br>No's |                           | Inspection (   | Requirements               | Status                 | Faults and/or R | emarks | Action Taken       |      | Initial |
| ALL<br>C                 |                           |  |                            |                        |                 |        |                    |      |         |
| ALL<br>C                 | 4.1.                      | Synchronized elevators for cracks in closing (inboard) rib with elevators removed.   |                            |                        |                 |        |                    |      |         |
| ALL<br>C                 | 5.                        | Bearings, bushings and rod ends in flight control linkages for excessive play and security. (Access panels 17, 18 and 36, Fig. 1-4). |                            |                        |                 |        |                    |      |         |
|                          |                           |  |                            |                        |                 |        |                    |      |         |
|                          |                           |  |                            |                        |                 |        |                    |      |         |

| РН                       | ASE | NO   | Area Name and No. TAILBOOM INTERIOR - 18 (CONT)   |        | A            | ircraft Serial No. | Date         |  |         |
|--------------------------|-----|--|---|--------|--------------|--------------------|--------------|--|---------|
| Inspect<br>Phase<br>No's |     | Inspection I   | Requirements  | Status | Fauils and/o | Remarks            | Action Taken |  | IrelMat |
| ALL<br>C                 | 6.  | tubes, links, bellc  | ages including push-pull<br>ranks, idlers, quadrant,<br>damage and security.<br>18 and 36, Fig. 1-4). |        |              |                    |              |  |         |
| ALL<br>C                 | 7.  | Tail rotor control cables for chafing broken wires and security. (Access panels 13, 14, 16, 17 and 18, Fig.1-4).                               |   |        |              |                    |              |  |         |
| ALL<br>C                 | 8.  | Tail rotor control cables for specified tension. (Access panel 17, Fig.1-4).   |   |        |              |                    |              |  |         |
| ALL<br>C                 | 9.  | Control cable pulleys for wear and damage. (Access panel 16, Fig.1-4).   |   |        |              |                    |              |  |         |
| ALL                      | 10. | Electrical wiring for chafing, deterioration and connector seals, and security of connections. (Access panels 14 thru 18, 36 and 83, Fig.1-4). |   |        |              |                    |              |  |         |

| Phase                   | e No TA   | Area Na<br>AIL ROTOR DRIVE                     |        |                 |         | Aircraft Serial No. | Dat         | 8       |
|-------------------------|---|--|--------|-----------------|---------|---------------------|-------------|---------|
| Inspect<br>Phase<br>No. | Inspection Require  | rements  | Status | Faults and/or R | lemarks | Action Taken        |             | Initial |
| ALL                     | 1. Intermediate (42°) gearbo  | ox oil drained, sight                          |        |                 |         |                     |             |         |
| С                       | gage for damage or stained glass, and refilled. (Access panel 13, Fig 1-4). |  |        |                 |         |                     |             |         |
|                         |   | (Necess paner 15, 1 ig 1 4).                   |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     | <del></del> |         |
|                         |   |  |        |                 |         |                     |             |         |
| ALL<br>C                | 2. Intermediate (42°) gearbo  | ox breather vent for                           |        |                 |         |                     |             |         |
|                         | for cracks, condition an panel 13, Fig 1-4).                                | nd security. (Access                           |        |                 |         |                     |             |         |
|                         | ранст 13, гід 1-4).   | }  |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     |             |         |
| ALL<br>C                | 3. Tail rotor control aft cables wires and security. (Access                | es for chaffing, broken test panels 13 and 14, |        |                 |         |                     |             |         |
|                         | Fig 1-4).   |  |        |                 |         |                     |             |         |
|                         |   | ŀ  |        |                 |         |                     |             |         |
|                         |   | ļ.   |        |                 |         |                     | <u>-</u>    |         |
| ALL                     | 4. Control cable pulleys for  | r wear and damage.                             |        |                 |         |                     |             |         |
| С                       | (Access panels 13 and 14,   | Fig 1-4).                                      |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     |             |         |
|                         | 5. Tail rotor driveshaft hang   | ger bearings. Inspect                          |        |                 |         |                     |             |         |
| С                       | IAW TM 55-1520-210-23<br>signed off if special inspe                        | ection was performed                           |        |                 |         |                     |             |         |
|                         | within 25 hours of current  | phase inspection).                             |        |                 |         |                     |             |         |
|                         |   |  |        |                 |         |                     |             |         |
|                         |   | i  |        |                 |         |                     |             | 1       |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

| Phase                   | No   | Area N<br>T/R AND GEARBOX                          | lame and No.<br>.— 20 |                 |         | Aircraft Serial No. | Date |         |
|-------------------------|--|--|-----------------------|-----------------|---------|---------------------|------|---------|
| Inspect<br>Phase<br>No. | Inspection R   | equirements  | Status                | Faults and/or F | Remarks | Action Taken        |      | Initial |
| ALL                     | 1. Vertical fin rib (P/N   | 1 204-030-827-27) along                            |                       |                 |         |                     |      |         |
| С                       | rivet row at fin station<br>thru topmost lightning<br>Fig. 1-4). | 10.08 for cracks (access holes). (Access panel 14, |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         | <u> </u>            |      |         |
|                         |  |  | }                     |                 |         |                     |      |         |
| ALL                     | Tail rotor (90°) gearbored for damage or stained                 | ox oil drained sight gage                          |                       |                 |         |                     |      |         |
| С                       | for damage or stained  | glass, and refilled.                               |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
| ALL                     | 3. Tail rotor (90°) gearb  | oox filler cap for clogged                         | ļ                     |                 |         |                     |      |         |
| С                       | vent.  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
| ALL                     | 4. Deleted.  |  |                       |                 |         | 1                   |      |         |
| С                       |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |
| ALL                     | 5. Tail rotor control roll damage and security.                  | ler chain and sprocket for                         |                       |                 |         |                     |      |         |
| С                       | damage and security. Fig. 1-4).                                  | (Access panels 14 and 29,                          |                       |                 |         |                     |      |         |
|                         | _  |  |                       |                 |         |                     |      |         |
|                         |  |  |                       |                 |         |                     |      |         |

"FOD REMINDER" Check work area for tools and part after completion of maintenance and inspection.

| Phase                   | e No 117   | Area N<br>R AND GEARBO                           | lame and No.<br>X - 20 (CO | NT)             | Aircr  | aft Serial No. | Da | ate     |
|-------------------------|--|--|----------------------------|-----------------|--------|----------------|----|---------|
| Inspect<br>Phase<br>No. | Inspection Requi   | rements  | Status                     | Faults and/or R | emarks | Action Taken   |    | Initial |
| ALL<br>C                | 6. Slowly operate tail rot<br>and observe roller chain<br>ensure no binding or clir<br>sprocket occurs (Access                             | operation to<br>nbing on the                     |                            |                 |        | <u></u>        |    |         |
| ALL<br>C                | 7. Remove tail rotor cont<br>for excessive grease on t<br>threads for wear. Thread<br>particles (Access Panel 2                            | ube. Splines and<br>ls for brass metal           |                            |                 |        |                |    |         |
| ALL                     | 8. Tail rotor control quill corrosion, leakage and so Panel 29, Fig. 1-4).   |  |                            |                 |        |                |    |         |
| ALL<br>C                | 9. Tail rotor control quill<br>splines which engage qu<br>on thread which engages<br>(pitch control assembly r<br>gearbox) (Access Panel 2 | ill housing and<br>s control nut<br>removed from |                            |                 |        |                |    |         |
| 3,6<br>C                | 10. Tail rotor (90°) gearb<br>(casting) for security and<br>chafing by vertical fin do   | l evidence of                                    |                            |                 |        |                |    |         |

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and inspection.

| Phase No T/R AND GEARBO |                       |               | ame and No.<br>X - 20 (CONT) |                 | А      | Aircraft Serial No. |         |
|-------------------------|-----------------------|---------------|------------------------------|-----------------|--------|---------------------|---------|
| Inspect<br>Phase<br>No. | Inspection I          |               | Status                       | Faults and/or R | emarks | Action Taken        | Initial |
| ALL                     | 11. Tail rotor asseml | bly balanced. |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       | İ             |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |
|                         |                       |               |                              |                 |        |                     |         |

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

| PH.                      | ASE | : NO   | Area N<br>OIL COOLER/AFT BA  |        |                 | A       | rcraft Serial No. | Date |         |
|--------------------------|-----|--|--|--------|-----------------|---------|-------------------|------|---------|
| Inspect<br>Phase<br>No's |     | Inspection F   | Requirements   | Status | Faults and/or F | Remarks | Action Taken      |      | Initial |
| 3,6                      | 1.  | . Remove oil cooler screens and clean fan blades and assembly. Check fan blades for cracks. (Access panel 36, Fig. 1-4). |  |        |                 |         |                   |      |         |
| ALL<br>C                 | 2.  | Check oil coolers for obstructions. Check turbine for rough or binding bearings by turning turbine by hand.              |  |        |                 |         |                   |      |         |
| ALL<br>C                 | 3.  | pull tubes, links,   | kages, including push-<br>bellcranks, bearings,<br>ends for excessive play,<br>and security. |        |                 |         |                   |      |         |
| ALL                      | 4.  | Electrical wiring for insulation and co  | or chafing, deterioration onnections.  |        |                 |         |                   |      |         |
| 6                        | 5.  |  | l tube. Check tube and<br>se rivets, cracks, corro-<br>n of holes.                           |        |                 |         |                   |      |         |

| L                        | HEATER COMPARTM |                                   | lame and   |               | A            | Aircraft Serial No.                   |              | е  |  |
|--------------------------|-----------------|-----------------------------------|--|---------------|--------------|---------------------------------------|--------------|----|--|
| Inspect<br>Phase<br>No's |                 | Inspection F                      | lequirements                                       | Status        | Faults and/o | r Remarks                             | Action Taken |    | Initial  |
| 3,6                      | 1.              | corrosion. (Access                | for damage, cracks and<br>panels 34 and 35, Fig.1- |               |              |                                       | ·            |    |  |
| 1                        |                 | 4).                               |  | <b>  </b>     |              |                                       |              |    |  |
| İ                        |                 |                                   |  | <del>  </del> |              |                                       |              |    | <b></b> -  |
|                          |                 |                                   |  |               |              |                                       |              |    | <b></b>  |
|                          | 2.              | Deleted.                          |  |               |              | <del>"</del>                          |              |    | <del> </del>                                     |
|                          |                 | Deleveu.                          |  |               |              |                                       |              |    | <del>                                     </del> |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
|                          | 1               |                                   |  |               | ····         |                                       |              |    |  |
|                          | i               |                                   |  |               |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              | -5 |  |
|                          |                 |                                   |  |               | ·            |                                       |              |    |  |
|                          |                 | DELETE                            |  |               |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
|                          | <u> </u>        |                                   |  |               |              |                                       |              |    |  |
|                          |                 | Deleted.                          |  |               |              | <del></del>                           |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              | · · · · · · · · · · · · · · · · · · · |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |
| 6                        | 5.              | Combustion heater                 | gnitor plug for condi-                             |               |              |                                       |              |    | -  |
|                          |                 | tion and security. (A. Fig. 1-4). | Access panels 34 and 35,                           |               | <del></del>  |                                       |              |    |  |
|                          |                 | r ig.1-4).                        |  |               | <del></del>  |                                       |              |    |  |
|                          |                 |                                   |  | <b></b>       |              |                                       |              |    |  |
|                          |                 |                                   |  |               |              |                                       |              |    |  |

|                          | PHASE NO. HEATER COMPARTM                                     |   | ame an |               | A       | Aircraft Serial No. D |  | •       |
|--------------------------|---|---|--------|---------------|---------|-----------------------|--|---------|
| Inspect<br>Phase<br>No's | Inspection (  | Requirements  | Status | Faults and/or | Remarks | Action Taken          |  | Initial |
| 6                        | 6. Combustion heat<br>assembly for cond<br>cess panels 34 and | er radiator and jacket<br>dition and security. (Ac-<br>35, Fig.1-4) |        |               |         |                       |  |         |
| 3,6                      | 7. Muff heater over<br>connector pins, cor                    | heat switch for loose<br>rosion and damage.                         |        |               |         |                       |  |         |
|                          |   |   |        |               |         |                       |  |         |
|                          | ·   |   |        |               |         |                       |  |         |
|                          | ·   |   |        |               |         |                       |  |         |

| 1                        | LUBRIC   | Area Name ar | nd No.          | A      | rcraft Serial No. | Dale    |
|--------------------------|--|--------------|-----------------|--------|-------------------|---------|
| Inspect<br>Phase<br>No s | Inspection Requirements  | Status       | Faults and/or R | emarks | Action Taken      | Initial |
| ALL<br>C                 | <ol> <li>Lubricate in accordance wit<br/>chart contained in the<br/>manual.</li> </ol> |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |
|                          |  |              |                 |        |                   |         |

| PHASE                    | NO  | POWER O | Area Name and No.     | Aircraft Serial No. | Date |        |
|--------------------------|---|---------|-----------------------|---------------------|------|--------|
| Inspect<br>Phase<br>No's | Inspection Requirements   | Status  | Faults and/or Remarks | Action Taken        |      | nitial |
| ALL<br>C                 | Cyclic and collective cylinders and connecting hydraulic lines for leaks. |         |                       |                     |      |        |
| ALL<br>C                 | Fuel lines for leaks during engine operation.                             |         |                       |                     |      |        |
| ALL<br>C                 | Tail rotor balanced, if not previously accomplished in Area 20.           |         |                       |                     |      |        |
| 3,6<br>C                 | Perform functional test on bleed air heater/muff heater system.           |         |                       |                     |      |        |
| ALL<br>C                 | Functional test windshield wiper motor and converter assembly.            |         |                       |                     |      |        |

## TM 55-1520-210-PM

| Phase No POWER ON CHECK |  | Area Name and<br>ON CHECKS (CO |                 |   | Aircraft Serial No. | Date |          |
|-------------------------|--|--------------------------------|-----------------|---|---------------------|------|----------|
| inspect<br>Phase<br>No. | Inspection Requirements  |                                | Faults and/or F | Remarks                                 | Action Taken        |      | Initial  |
| i                       | ſ  | tity indicator                 |                 |   |                     |      |          |
| 3, 6<br>C               | 6. Perform inspection on fuel quant and 20 minute fuel caution light readings. | t for correct                  |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 | *************************************** |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      | <u> </u> |
|                         |  |                                |                 |   |                     |      |          |
|                         |  | ļ                              |                 |   |                     |      |          |
|                         |  |                                |                 |   |                     |      |          |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

| Phase No FINAL INSPECTION |  |   |              |                 |                                       | Aircraft Serial No. Da |             | e       |
|---------------------------|--|---|--------------|-----------------|---------------------------------------|------------------------|-------------|---------|
| igepect<br>Phase<br>No.   | Inspection Requirements  |   | Status       | Faults and/or F | temarks                               | Action Taken           |             | Initial |
| ALL<br>C                  |  |   |              |                 |                                       |                        |             |         |
|                           |  |   |              |                 |                                       |                        |             |         |
|                           |  |   |              |                 |                                       |                        |             |         |
| ALL<br>C                  |  |   |              |                 |                                       |                        |             |         |
|                           |  |   |              | ;               | · · · · · · · · · · · · · · · · · · · |                        | <del></del> |         |
| ALL<br>C                  |  |   |              |                 |                                       |                        |             |         |
|                           |  |   |              |                 |                                       |                        |             |         |
|                           |  | · |              |                 |                                       |                        |             |         |
|                           |  |   |              |                 | <del></del>                           |                        | <del></del> |         |
| ALL<br>C                  | 3. Release aircraft from inspection status to permit accomplishment of post inspection maintenance test flight (MTF) in accordance with requirements of TM 55-1520-242-MTF and TM 1-1500-328-23. |   |              |                 |                                       |                        |             |         |
|                           |  |   |              |                 |                                       |                        |             |         |
|                           |  |   |              | •               |                                       |                        |             |         |
|                           |  |   |              |                 |                                       |                        |             | ļ       |
|                           |  |   |              |                 |                                       |                        |             |         |
|                           |  |   | <b></b>      |                 |                                       |                        | ··          |         |
|                           |  |   | <del> </del> |                 |                                       |                        |             | <b></b> |
|                           |  |   |              |                 |                                       |                        |             |         |

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

By Order of the Secretary of the Army:

E. C. MEYER General, United State Army Chief of Staff

Official:

ROBERT M. JOYCE Major General, United States Army The Adjutant General

## **DISTRIBUTION:**

To be distributed in accordance with DA Form 12-31, PM requirements for UH-1D/H and EH-1H aircraft.

☆U. S. GOVERNMENT PRINTING OFFICE: 1989 0 - 242-894 (7072)

The following format must be used if submitting an electronic 2028. The subject line must be exactly the same and all fields must be included; however only the following fields are mandatory: 1, 3, 4, 5, 6, 7, 8, 9, 10, 13, 15, 16, 17, and 27.

From: "Whomever" < whomever@avma27.army.mil>

To: <ls-lp@redstone.army.mil>

Subject: DA Form 2028

1. *From:* Joe Smith

2. Unit: home

Address: 4300 Park
 City: Hometown

5. **St**: MO6. **Zip**: 77777

7. Date Sent: 19-OCT-93
 8. Pub no: 55-2840-229-23

9. **Pub Title:** TM

10. Publication Date: 04-JUL-85

Change Number: 7
 Submitter Rank: MSG
 Submitter F Name: Joe
 Submitter MI Name: T

15. Submitter L Name: Smith

16. **Submitter Phone:** 123-123-1234

17. **Problem: 1**18. Page: 2
19. Paragraph: 3
20. Line: 4

20. Line. 4 21. NSN: 5

22. Reference: 6

23. Figure: 7 24. Table: 8 25. Item: 9 26. Total: 123

27. **Text:** This is the text for the problem below line 27.



# **SOMETHING WRONG**

WITH THIS PUBLICATION?

THEN . .JOT DOWN THE DOPE ABOUT IT ON THIS FORM, CAREFULLY TEAR IT OUT, FOLD IT AND DROP IT IN THE MAIL!

FROM: (PRINT YOUR UNIT'S COMPLETE ADDRESS) PFC John DOE CO A 3rd Engineer Bn Ft. Leonardwood, MO 63108

DATE SENT

22 August 1992

PUBLICATION NUMBER TM 55-1520-210-PM PUBLICATION DATE 15 May 19%

**PUBLICATION TITLE** Phased Maintenance Checklist, UH-1H/V and EH-1H/X Aircraft

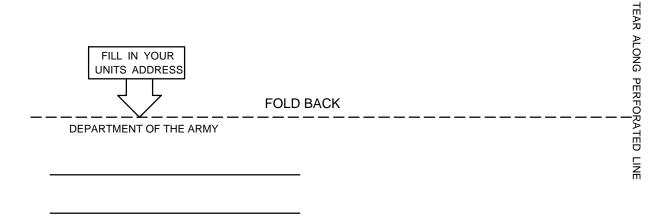
| PRESENCE PILAPOINT WHERE IT IS  PAGE PARA FIGURE TABLE NO 2-1 a In line 6 of paragraph 2-1a the manual states the engine has 6 cylinders. The engine on my set only has 4 cylinders.  Change the manual to show 4 cylinders.  Callout 1.6 on figure 4-3 is pointed at a bolt. In key to figure 4-3, item 16 is calle a shim Please correct one or the other  |                                  | and ETFTT/A Allclait  |  |  |  |  |  |  |
|--|----------------------------------|---|--|--|--|--|--|--|
| So GRAPH NO NO So So Solve Sol | BE EXACT PIN-POINT WHERE IT IS   | IN THIS SPACE, TELL WHAT IS WRONG  AND WHAT SHOULD BE DONE ABOUT IT:  |  |  |  |  |  |  |
| the manual states the engine has 6 cylinders. The engine on my set only has 4 cylinders. Change the manual to show 4 cylinders.  Callout 16 on figure 4-3 is pointed at a bolt. In key to figure 4-3, item 16 is calle a shim. Please correct  | 1,7,5,5                          |   |  |  |  |  |  |  |
|  | 6 2-1 a                          | the manual states the engine has 6 cylinders. The engine on my set only has $\frac{4}{4}$ cylinders. Change the manual to show 4 cylinders.  Callout 16 on figure 4-3 is pointed at a bolt. In key to figure 4-3, item 16 is calle a shim. Please correct |  |  |  |  |  |  |
| PRINTED NAME, GRADE OR TITLE, AND TELEPHONE NUMBER  SIGN HERE  | PRINTED NAME, GRADE OR TITLE, AN |   |  |  |  |  |  |  |
| JOHN DOE, PFC (268) 317-7111 JOHN DOE  | JOHN DOE, PFC (268) 317-         | JOHN DOE, PFC (268) 317-7111 JOHN DOE   |  |  |  |  |  |  |

DA FORM 2028-2

PREVIOUS EDITIONS ARE OBSOLETE.

DRSTS-M verprint2, 1 Nov 80

P.S.- - IF YOUR OUTFIT WANTS TO KNOW ABOUT YOUR RECOMMENDATION, MAKE A CARBON COPY OF THIS AND GIVE TO YOUR HEADQUARTERS.



**OFFICIAL BUSINESS** 

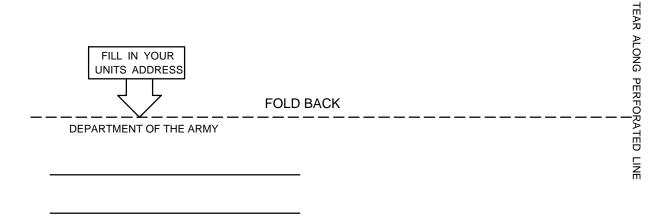
Commander USAAMCOM

ATTN: AMSAM-MMC-LS-LP

Redstone Arsenal, AL 35898-5230

|   |                                 | SOMETHI  | NG '     | WRONG WITH THIS PUBLICATION?                |  |  |
|---|---------------------------------|--|----------|---|--|--|
|   | DOPE ABO<br>FORM, CA<br>IT OUT, | OT DOWN THE<br>OUT IT ON THIS<br>REFULLY TEAR<br>FOLD IT AND<br>IN THE MAIL! | FROM:    | : (PRINT YOUR UNIT'S COMPLETE ADDRESS) SENT |  |  |
| PUBLICATION NUMBER PUBLICATION DATE PUBLICATION TITLE |                                 |  |          |   |  |  |
| PAGE PARA-NO GRAPH NO                                 | RE TABLE NO                     | THIS SPACE, TELL WED WHAT SHOULD BE  | AAT IS W | IRONG BOUT IT:                              |  |  |
| PRINTED NAME, GRADE                                   | OR TITLE, AND TEL               | LEPHONE NUMBER   | SIG      | GN HERE                                     |  |  |

DA FORM 2028-2



**OFFICIAL BUSINESS** 

Commander USAAMCOM

ATTN: AMSAM-MMC-LS-LP

Redstone Arsenal, AL 35898-5230

# The Metric System and Equivalents

### Linear Measure

1 centimeter = 10 millimeters = .39 inch 1 decimeter = 10 centimeters = 3.94 inches 1 meter = 10 decimeters = 39.37 inches 1 dekameter = 10 meters = 32.8 feet

1 hectometer = 10 dekameters = 328.08 feet

1 kilometer = 10 hectometers = 3,280.8 feet

#### Weights

1 centigram = 10 milligrams = .15 grain 1 decigram = 10 centigrams = 1.54 grains 1 gram = 10 decigram = .035 ounce 1 dekagram = 10 grams = .35 ounce 1 hectogram = 10 dekagrams = 3.52 ounces 1 kilogram = 10 hectograms = 2.2 pounds 1 quintal = 100 kilograms = 220.46 pounds 1 metric ton = 10 quintals = 1.1 short tons

### Liquid Measure

1 centiliter = 10 milliters = .34 fl. ounce 1 deciliter = 10 centiliters = 3.38 fl. ounces 1 liter = 10 deciliters = 33.81 fl. ounces 1 dekaliter = 10 liters = 2.64 gallons 1 hectoliter = 10 dekaliters = 26.42 gallons 1 kiloliter = 10 hectoliters = 264.18 gallons

### Square Measure

1 sq. centimeter = 100 sq. millimeters = .155 sq. inch 1 sq. decimeter = 100 sq. centimeters = 15.5 sq. inches 1 sq. meter (centare) = 100 sq. decimeters = 10.76 sq. feet 1 sq. dekameter (are) = 100 sq. meters = 1,076.4 sq. feet 1 sq. hectometer (hectare) = 100 sq. dekameters = 2.47 acres 1 sq. kilometer = 100 sq. hectometers = .386 sq. mile

### Cubic Measure

1 cu. centimeter = 1000 cu. millimeters = .06 cu. inch 1 cu. decimeter = 1000 cu. centimeters = 61.02 cu. inches 1 cu. meter = 1000 cu. decimeters = 35.31 cu. feet

# **Approximate Conversion Factors**

| To change     | To                 | Multiply by | To change          | To            | Multiply by |
|---------------|--------------------|-------------|--------------------|---------------|-------------|
| inches        | centimeters        | 2.540       | ounce-inches       | newton-meters | .007062     |
| feet          | meters             | .305        | centimeters        | inches        | .394        |
| yards         | meters             | .914        | meters             | feet          | 3.280       |
| miles         | kilometers         | 1.609       | meters             | yards         | 1.094       |
| square inches | square centimeters | 6.451       | kilometers         | miles         | .621        |
| square feet   | square meters      | .093        | square centimeters | square inches | .155        |
| square yards  | square meters      | .836        | square meters      | square feet   | 10.764      |
| square miles  | square kilometers  | 2.590       | square meters      | square yards  | 1.196       |
| acres         | square hectometers | .405        | square kilometers  | square miles  | .386        |
| cubic feet    | cubic meters       | .028        | square hectometers | acres         | 2.471       |
| cubic yards   | cubic meters       | .765        | cubic meters       | cubic feet    | 35.315      |
| fluid ounces  | milliliters        | 29,573      | cubic meters       | cubic yards   | 1.308       |
| pints         | liters             | .473        | milliliters        | fluid ounces  | .034        |
| quarts        | liters             | .946        | liters             | pints         | 2.113       |
| gallons       | liters             | 3.785       | liters             | quarts        | 1.057       |
| ounces        | grams              | 28.349      | liters             | gallons       | .264        |
| pounds        | kilograms          | .454        | grams              | ounces        | .035        |
| short tons    | metric tons        | .907        | kilograms          | pounds        | 2.205       |
| pound-feet    | newton-meters      | 1.356       | metric tons        | short tons    | 1.102       |
| pound-inches  | newton-meters      | .11296      |                    |               |             |

# Temperature (Exact)

°F Fahrenheit temperature

5/9 (after subtracting 32) Celsius temperature °C

PIN: 017672-000