

24/ARMY

STUZEXN RULNEAA079 2630111-UUUU-RUMMDTA;
VR UUUUU
032320Z MAR 83 ZEX
1 MGR ARNG OAC EA APG MD //NGB-AVN-L//
1 AIG 7401
30T DA-BHCSVD
032000Z MAR 83
1 CDRTSARCOM STL MO //DRSTS-MEA//

3 AIG 8881
JEMKY/CHMDO TOKYO JAPAN
JCLRFA/AMERICAN ELECTRONICS LABORATORY, PO BOX 691,
FARMINGDALE, NJ 07727//
NFO AIG 9004
IG 7515
JCLRFA/SANDERS ASSOCIATES INC, 95 CANALST, MASHUA, NH 03061
/DAMA-HSA FOR COL H. MALONEY//
JCLRFA/KAMAN AEROSPACE CORP OLD WINDSOR RD BLOOMFIELD CT
/26002//
JQHAD/USDAO TEL AVIV ISRAEL
JLRFA/US ENV PROT AGY REGION 11 26 FEDERAL PLAZA NY NY
JOLMA/NATIONAL OCEANIC AND ATMOSPHERIC ADM ROCKVILLE MD
/NOAA.OA2 COL L. A. HEAVER//
JOLEUA/TSM ATK HEL FT RUCKER AL //ATZQ-TSM-A//
JEHC/AIT WASH DC
JFLAFA/CJUSHAG MAAG MADRID SPAIN //ARMY/NAVY SEC//
JCLRFA/DOI DEF MISS 850 3D AVE NY NY 10022

UH-1-83-04
C/M models
03337

AGE 02 RUCIFRA UNCLAS
30T DA-BHCSVD

UNCLAS

ACTION 01 OF 05
1JGR GENERAL EMIL L. KONOPNICKI SENDS
JR NAS JACKSONVILLE FL PASS TO CUSTOMS AIR BRANCH
JR HOMESTEAD FL PASS TO US CUSTOMS AIR BRANCH BLDG 200
SUBJECT SAFETY-OF-FLIGHT TECHNICAL MESSAGE, ONE TIME INSPECTION FOR
CERTAIN 540 SERIES MAIN ROTOR BLADES (NSN 1615-20-389-1950) ON ALL
UH-1 (C AND M ONLY) AND AH-1 SERIES AIRCRAFT (AH-1-83-02) AND
UH-1-83-04 (C AND M MODELS ONLY)
NOTE THIS IS A SAFETY-OF-FLIGHT, TECHNICAL, ONE-TIME INSPECTION
MESSAGE, RELEASED IN ACCORDANCE WITH AR 95-18 DATED 1 MAY 1980 AND
AS NOT, REPEAT HAS NOT, BEEN TRANSMITTED TO UNITS SUBORDINATE TO
ADDRESSEES. ADDRESSEES SHOULD IMMEDIATELY RETRANSMIT THIS MESSAGE TO
SUBORDINATE UNITS, ACTIVITIES, OR ELEMENTS AFFECTED OR CONCERNED.
RE RETRANSMITTAL SHALL REFERENCE THIS MESSAGE; ACTION ADDRESSEES
MUST IMMEDIATELY VERIFY THIS RETRANSMISSION TO COMMANDER, TSARCOM,

(8)

IN DRSTS-MPSD,
TSARCOM, DRSTS-MEA, 252305Z FEB 83, SUBJECT SAFETY-OF-FLIGHT
EMERGENCY MESSAGE, ONE TIME INSPECTION FOR CERTAIN 540 SERIES MAIN

AGE 03 RUCIFRA5345 UNCLAS
ROTOR BLADES ON ALL UH-1 (C AND H ONLY) AND AH-1 SERIES AIRCRAFT
(AH-1-83-01) AND (UH-1-83-03) (C AND H MODELS ONLY)

SUMMARY

A. RECENTLY A PRODUCTION RUN OF REPAIRED MAIN ROTOR BLADES WERE
ISSUED TO THE USER. SUBSEQUENTLY, IT WAS DETERMINED THAT BLADE

REPAIR PROCEDURES BY THE REPAIR CONTRACTOR WERE DEVIATED FROM ON
REPAIRED BLADES. THIS COULD RESULT IN DEFECTIVE BLADES.
ACCORDINGLY, TSARCOM ISSUED REFERENCE A MESSAGE GROUNDING ALL BLADES
IN AFFECTED LOT UNTIL AN INSPECTION COULD BE DEVELOPED THAT WOULD
ENSURE SERVICEABILITY OF SUSPECT BLADES.

B. THE FOLLOWING REPAIR PROCEDURES WERE SUSPECT

(1) TORQUE VALUES FOR TIP AND BUTT (ROOT) BOLTS ATTACHING
BALANCE WEIGHTS WERE FOUND TO BE IMPROPER. LOW TORQUE VALUES COULD
RESULT IN LOOSE NUTS AND A SUBSEQUENT DANGEROUS ROTOR OUT OF BALANCE
CONDITION. THIS MESSAGE PROVIDES INSTRUCTIONS FOR INSPECTING ROOT
AND TIP NUTS AND RETORQUING TO REQUIRED TORQUE SETTINGS (PARAGRAPH 4
BELOW).

(2) BONDING INTEGRITY OF LEADING EDGE EROSION GUARD IS
QUESTIONABLE. ONLY THREE 540 BLADES HAD LEADING EDGE EROSION GUARD
BONDED ON BY CONTRACTOR. THE THREE 540 BLADES ARE TO BE REMOVED AND

AGE 04 RUCIFRA5345 UNCLAS
REPLACED IN ACCORDANCE WITH THE INSTRUCTIONS IN THIS MESSAGE
(PARAGRAPH 2 BELOW).

C. THE ABOVE PROCEDURES HAVE BEEN CHECKED AND SUSPECT BLADES ARE
EITHER ACCEPTABLE, REQUIRE INSPECTION, REWORK, OR REMOVAL FROM
SERVICE IN ACCORDANCE WITH INSTRUCTIONS IN THIS MESSAGE.

D. THIS SAFETY-OF-FLIGHT MESSAGE PROVIDES INSTRUCTIONS FOR THE
INSPECTION AND RETORQUING TO BE PERFORMED ON ROOT AND TIP BALANCE
EIGHT NUTS OF SUSPECT BLADES (SERIAL NUMBERS LISTED IN PARAGRAPH 3
BELOW). IT ALSO PROVIDES INSTRUCTIONS FOR REMOVING FROM SERVICE THE
THREE 540 BLADES THAT HAD LEADING EDGE GUARDS BONDED ON DURING BLADE
REPAIR.

ONLY THREE 540 BLADES HAD THE LEADING EDGE EROSION GUARD REPLACED
AND THEY MUST BE TURNED INTO SUPPLY FOR REPAIR. THE 540 BLADE SERIAL
NUMBERS ARE A2-58369 (WHICH IS CURRENTLY AT CCAD UNDERGOING A
WEAR/DOWN ANALYSIS), A2-28633 AND A2-56140.

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0279

ZA / ARMY

411111 04 01 2

BTUZEKX RULNEAA0000 0630112-0000--RUKMDTA,
NR 00000
032320Z MAR 83 ZEX
'M MGR ARNG OAC EA APG MD //NGB-AVN-L//
'O AIG 7421
'COT DA-BHCSVD
' 032000Z MAR 83
'M CDRTSARCOM STL MD //DRSTS-MEA//

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'EOLMA/NATIONAL OCEANIC AND ATMOSPHERIC ADM ROCKVILLE MD
'/NOAA, OAS COL L. A. HEAVER//
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'AGE 02 RUCIFRA UNCLAS
'COT DA-BHCSVD
BT
INCLAS

SECTION 02 OF 05
1. THE SERIAL NUMBERS OF THE 540 SERIES BLADES REQUIRING BALANCE
WEIGHT INSPECTION ARE PROVIDED BELOW

HR-00031	A2-4087	A2-09383
-50014	-4708	-09398
-50024	-4867	-09434
-50030	-4949	-09450
-50120	-5138	-09477
-50148	-5491	-09479
-50156	-5995	-09481
-50185	-6527	-09708
-50198	-7995	-09874
-50201	-8218	-10101
-50205	-84667	-10348
-50274	-85627	-10367

R-58289
-58316
-58329

A2 -58439
-58515
-58518

A2 -58427
-58332
-58383

AGE 23 RUCIFRA5347 UNCLAS

-58464	-58689	-581447
-58482	-58827	-58811
-58482	-58885	-58888
-58491	-58112	-58874
-58554	-58137	-58884
-58561	-58246	-58122
-58585	-58297	-58134
-58613	-58315	-58148
-58645	-58339	-58158
-58658	-58346	-58152
-58788	-58349	-58154
-58794	-58361	-58167
-58836	-58375	-58187
-58879	-58407	-58181
-57241	-58425	-58369
-57273	-58481	-58398
HR-57353	A2-58493	A2-58647
-57355	-58552	-58835
-57454	-58689	-58855
-57583	-58833	-58871

AGE 24 RUCIFRA5347 UNCLAS

-57687	-58639	-58121
Z-421	-58641	-58136
-1883	-58641	-588843
-1195	-58644	A//58874
-1548	-58718	1HB-5917
-1824	-58744	-2626
-2355	-58746	-2858
-2834	-58769	-3275
-3692	-58774	12-843
-3939	-59368	

INSPECTION OF BLADE BALANCE HEIGHTS
REFERENCE AVIATION UNIT AND INTERMEDIATE MAINTENANCE REPAIR PARTS
AND SPECIAL TOOLS LIST
A. TM 55-1528-221-23P-1, 26 DEC 79, PAGE 336, FIGURE 82,
B. TM 55-1528-234-23P-1, 12 MAR 82, PAGE 298, FIGURE 75,
C. TM 55-1528-236-23P-1, 3 MAY 82, PAGE 354, FIGURE 64,
D. TM 55-1528-239-23P-1, 29 JAN 80, PAGE 388, FIGURE 65,
E. TM 55-1528-218-23P-1, 27 FEB 81, PAGE 658, FIGURE 136 (FOR
48 BLADES).
GENERAL INSTRUCTIONS

GE 05 RUCIFRA3347 UNCLAS

A. REFERENCES LISTED ABOVE ARE THE APPLICABLE PARTS MANUALS THAT PERTAIN TO THE 340 SERIES BLADES IN QUESTION. THE APPROPRIATE FIGURE/PAGE NUMBERS ARE PRESENTED FOR QUICK REFERENCE.

B. THE BLADES MUST BE REMOVED FROM THE MAIN ROTOR HUB TO FACILITATE MAINTENANCE INSPECTION REQUIREMENTS.

C. REFER TO PARAGRAPH 9 FOR THE INSPECTION ITEM IDENTIFICATION NUMBER, CROSS REFERENCE WITH THE ITEM NUMBER FROM THE APPROPRIATE PARTS MANUAL.

D. REFER TO PARAGRAPH 10 FOR THE INSPECTION ITEM NUMBER CROSS REFERENCE TO THE PART NUMBER LISTED IN THE PARTS MANUAL FOR THE SPECIFIED NUT, SCREW, AND BOLT INSTALLATION TORQUE VALUES.

E. SINCE THE WEIGHT INSTALLATION IS REQUIRED TO BE PERFORMED AT SPOT LEVEL, IT IS RECOMMENDED THAT THE LOCAL TSARCOM FMT BE CONSULTED CONCERNING ANY PROBLEMS/QUESTIONS.

3030

24 / ARMY

OTUZEXM RULNEAA8881 8638114-UUUU--RUHMOTA,
NR UUUUU

032328Z MAR 83 ZEX
1 MGR ARNG OAC EA APG MD //NGB-AVN-L//
1 AIG 7484

10T DA-BHCSVD
832888Z MAR 83
1 ODRTSARCOM STL MO //DRSTS-MEA//

1 AIG 8881
JEHXY/CHMDO TOKYO JAPAN
JCLRFA/AMERICAN ELECTRONICS LABORATORY, PO BOX 691,
'FARMINGDALE, NJ 07727//
1FO AIG 9884

1G 7515
JCLRFA/SANDERS ASSOCIATES INC, 95 CANALST, MASHUA, NH 03861
'DAMA-WSA FOR COL H. MALONEY//
JCLRFA/KAMAN AEROSPACE CORP OLD WINDSOR RD BLOOMFIELD CT
'86882//

JOMAD/USDAO TEL AVIV ISRAEL
JCLRFA/US ENV PROT AGY REGION 11 26 FEDERAL PLAZA NY NY
'EOLMA/NATIONAL OCEANIC AND ATMOSPHERIC ADM ROCKVILLE MD
'NOAA, O2 COL L. A. WEAVER//
JOLEUA/TSM ATK HEL FT RUCKER AL //ATZG-TSM-A//
JEHC/AIT WASH DC
JFLAFA/OJUSMAG MAAG MADRID SPAIN //ARMY/NAVY SEC//
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4 Mar 83 04 06z

10E 22 RUCIFRA UNCLAS
10T DA-BHCSVD

UNCLAS

SECTION 23 OF 25

F. WHEN REMOVAL OF THE WEIGHT RETENTION NUTS IS REQUIRED, DO NOT
MOVE THE BALANCE WEIGHTS,

TIP WEIGHTS,

A. REMOVE SCREWS (2A AND 2B) FROM TIP CAP PLATE (1) (3 SCREWS,
1T SAME SIZE),

B. REMOVE TIP CAP PLATE (1), THIS MAY REQUIRE SOME HEAT (FROM
HOT GUN) AND LIGHT PRYING TO BREAK TIP CAP PLATE SEALANT,

C. INSPECT TIP WEIGHTS FOR PROPER INSTALLATION,
(1) CHECK CLOCKWISE BREAKAWAY TORQUE ON TIP WEIGHT RETENTION

1T (3).

2. BREAKAWAY TORQUE EXCEEDS 225 INCH POUNDS, BLADES WILL BE REMOVED
FROM AIRCRAFT AND TURNED INTO SUPPLY (SUBMIT QDR PER TM 38-758).

NOTE IF THE STANDARD THREE INCH DEEPWELL SOCKET IS NOT DEEP ENOUGH
TO ENGAGE THE NUT, IT MAY BE NECESSARY TO PURCHASE OR FABRICATE A

EPER SOCKET,

(2) IF CLOCKWISE BREAKAWAY TORQUE IS LESS THAN THE MINIMUM SPECIFIED IN PARAGRAPH 10, RETORQUE RETENTION NUT (3) TO THE

AGE 03 RUCIFRA5349 UNCLAS

APPROPRIATE LEVEL IN PARAGRAPH 10,

(3) IF CLOCKWISE BREAKAWAY TORQUE EXCEEDS THE MAXIMUM SPECIFIED IN PARAGRAPH 10 BUT IS BELOW 225 INCH POUNDS, REMOVE THE ORIGINAL RETENTION NUT (3) AND INSTALL NEW NUTS TORQUED TO THAT SPECIFIED IN PARAGRAPH 10.

(4) REINSTALL TIP CAP PLATE AND RETENTION SCREWS; TIGHTEN SCREWS (2A AND 2B) EVENLY PER THE TORQUE VALUES IN PARAGRAPH

NOTE DO NOT APPLY PROSEAL TO TIP CAP PLATE AND SCREWS;

BUTT (ROOT) WEIGHTS

A. REMOVE BUTT COVER RETENTION SCREWS (5A AND 5B) FROM BUTT COVER PLATE (4).

CAUTION

THE FOUR WEIGHT RETENTION BOLTS (6) ARE THROUGH BOLTS MOUNTED TO THE BUTT COVER PLATE. SEE PARAGRAPH 9 BELOW FOR PROPER TECHNICAL MANUAL TO IDENTIFY THE BUTT COVER PLATE RETENTION SCREWS AND THE WEIGHT RETENTION BOLTS.

B. REMOVE THE BUTT COVER PLATE AND WEIGHTS. THIS MAY REQUIRE THE HEAT (FROM HEAT GUN) AND LIGHT PRYING TO BREAK THE BUTT COVER

AGE 04 RUCIFRA5349 UNCLAS

LATE SEALANT.

C. INSPECT WEIGHTS FOR PROPER INSTALLATIONS;

(1) CHECK CLOCKWISE BREAKAWAY TORQUE ON THE BUTT WEIGHT RETENTION BOLTS (6) AND NUTS (7). IF CLOCKWISE BREAKAWAY TORQUE EXCEEDS 225 INCH POUNDS, BLADES WILL BE REMOVED FROM AIRCRAFT AND TURNED INTO SUPPLY. (SUBMIT QDR PER TM 38-750).

(2) IF CLOCKWISE BREAKAWAY TORQUE EXCEEDS THE MAXIMUM SPECIFIED IN PARAGRAPH 10 BUT IS BELOW 225 INCH POUNDS, REMOVE THE ORIGINAL RETENTION NUTS (7) AND INSTALL NEW NUTS TORQUED TO THAT SPECIFIED IN PARAGRAPH 10.

(3) IF CLOCKWISE BREAKAWAY TORQUE IS LESS THAN THE MINIMUM SPECIFIED, RETORQUE RETENTION NUT (7) TO THE APPROPRIATE LEVEL IN PARAGRAPH 10.

(4) REPLACE BUTT COVER PLATE AFTER THOROUGHLY CLEANING THE SURFACE COAT MATING SURFACES WITH A THIN COAT OF SEALING COMPOUND MIL-S-8882. REINSTALL BUTT COVER PLATE, WEIGHTS AND MOUNTING SCREWS. APPLY A RING OF SEALING COMPOUND MIL-S-8882 UNDER HEADS OF SCREWS. ONLY TIGHTEN SCREWS (5A AND 5B) PER THE TORQUE VALUES IN PARAGRAPH

SPAN WEIGHTS INSPECT SPAN WEIGHTS (8) FOR PROPER INSTALLATION.

PAGE 05 RUCIFRA5349 UNCLAS

TORQUE SPAN WEIGHT RETENTION SCREWS (9) TO THAT SPECIFIED IN
PARAGRAPH 10,

INSPECTION ITEM/PART MANUAL ITEM NO, CROSS REFERENCE PARTS MANUAL
1 55-1520-XXX-23P1,

ITEM	NOMENCLATURE	-221	-234	-236	-239	-210 (540)
TIP CAP WEIGHTS						
	TIP CAP PLATE	3	3	3	3	31
	TIP CAP PLATE RET SCREW A					
		1	1	1	1	29
	TIP CAP PLATE RET SCREW B					
		2	2	2	2	30

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2A/ARMY

BTUZEKX RULNEAA0082 0630122-~~UUUU~~-RUMMDTA;
NR UUUUU
032320Z MAR 83 ZEX
1 MOR ARNO OAC EA APG MD //NGB-AVN-L//
1 AIG 7401
JOT DA-BHCSVD
032000Z MAR 83
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JCLRFA/US SBY PROT AGY REGION 11 26 FEDERAL PLAZA NY NY
JECOLMA/NATIONAL OCEANIC AND ATMOSPHERIC ADM ROCKVILLE MD
NDAA, OAZ COL L. A. WEAVER//
JLEUA/TSM ATK HEL FT RUCKER AL //ATZQ-TSM-A//
JEMC/AIT WASH DC
JFLAFA/CJUSHAG MAAG MADRID SPAIN //ARMY/NAVY SEC//
JCLRFA/GOI DEF MISS 850 3D AVE NY NY 10022

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1GE 02 RUCIFRA UNCLAS
JOT DA-BHCSVD

UNCLAS

SECTION 24 OF 25

TIP WGT RET NUT 4 4 4 4 32

ITT WEIGHTS

BUTT COVER PL	43	43	29	28	25
BUTT COVER RET SCREW A	39	39	25	24	23
BUTT COVER RET SCREW B	41	41	27	26	21
BUTT WEIGHT RET BOLT	45	45	38	29	20

BUTT WEIGHT RET NUT
 44 44 4 4 11

SPAN WEIGHTS

SPAN WEIGHT 31-38 31-38 24 23 28

SPAN WEIGHT RET SCREW A
 28 28 22 21 26

SPAN WEIGHT RET SCREW B
 29-38 29-38 23 22 27

AGE 23 RUCIFRA5351 UNCLAS

3. INSPECTION ITEM/NSN/INSTALLATION TORQUE (IN POUNDS)

ITEM	NOMENCLATURE	BLADE	P MANUAL	NSN	TORQUE RANGE
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	TIP CAP PLATE	548	ALL	548-811-825-1	NA
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1	TIP CAP PLATE (RETENTION SCREW)	548	ALL	NK 589-18R18	12-15
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3	TIP CAP PLATE (RETENTION SCREW)	548	ALL	NK 589-416R18	38-48
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	TIP WEIGHT (RETENTION NUT)	548	-221/-234	MS 21842L3	138-145
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		548	-236/-239	MS 21842-5	
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		548	-218	MS 21842L5	
--	--	-----	------	------------	--

	BUTT COVER	548	ALL	548-811-817-1	NA
--	------------	-----	-----	---------------	----

1	BUTT COVER (RETENTION SCREW)	548	-221/234 -239/-218	NAS 1198-3P8L	12-15
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	BUTT COVER (RETENTION SCREW)	548	-236	NK 525-18RB	
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3	BUTT COVER (RETENTION SCREW)	548	ALL	NK 525-416-RB	38-48
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	BUTT WEIGHT (RETENTION BOLT)				
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AGE 24 RUCIFRA5351 UNCLAS

		548	ALL	NAS 428-5-34	138-145
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	BUTT WEIGHT (RETENTION NUT)				
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540

COBRA

MS 21042-5

130-145

540

UH-1

MS 2104-2L3

130-145

SPAN WEIGHT

540

ALL

540-011-019-1,
3,5,9,11,13,OR 15 NA

A SPAN WEIGHT (RETENTION SCREW)

540

=221/234

=239/210

NAS 1635-3LE8

12-15

540

=236

NAS 1635-3LE8

12-15

B SPAN WEIGHT (RETENTION SCREW)

T
0082

NNN

DTUZEXW RULNEAAZ083 2638122(UUUU)-RUKMDTA;

NR UUUUU

032320Z MAR 83 ZEX

M MGR ARNG OAC EA APG MD //NGB-AVN-L//

C AIG 7481

CCT DA-BHCSVD

032000Z MAR 83

M CDRTSARCOM STL MO //DRSTS-HEA//

27 / Army

O AIG 8881

UEHKY/CHMDO TOKYO JAPAN

UCLRFA/AMERICAN ELECTRONICS LABORATORY, PO BOX 691,

/FARMINGDALE, NJ 07727//

VFO AIG 9884

IG 7513

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/26202//

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/NOAA, OAS COL L. A. HEAVER//

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JEMO/AIT WASH DC

JFLAFA/OJUSHAG MAAG MADRID SPAIN //ARMY/NAVY SEC//

JCLRFA/GOI DEF MISS 850 3D AVE NY NY 10822

411073 04102

~~AGE 22 RUCIFRA UNCLAS~~

~~CCT DA-BHCSVD~~

~~T~~

~~NCLAS~~

~~INAL SECTION OF 03~~

548	COBRA	NAS 1198-3P8L OR -3P10L	12-15
548	UH-1	NAS 1198-3P6L OR 8L, 10L, 12L, 14L, 18L, 20L, 22L, 24L, 26L	12-15

LLI REFERS TO -221, -234, -236, -239, AND -218 MANUALS FOR THE 548

LADE.

1. ONCE THE SUBJECT BLADES HAVE BEEN SUBJECTED TO THE REQUIREMENTS OF THIS MESSAGE AND SATISFY THE LOCAL TECHNICAL INSPECTOR, THE BLADES CAN BE RELEASED FOR FLIGHT.

2. RECORDING AND REPORTING OF INSPECTION

A. RECORD ACCOMPLISHMENT OF YOUR INSPECTION IN ACCORDANCE WITH

THE PROCEDURE IN TM 38-752, THE FOLLOWING FORMS ARE APPLICABLE,
(1) DA FORM 2488-13, AIRCRAFT INSPECTION AND MAINTENANCE
RECORD.

- (2) DA FORM 2488-25, AIRCRAFT HISTORICAL RECORD,
- (3) DA FORM 2488-16, AIRCRAFT COMPONENT HISTORICAL RECORD

AGE 23 RUCIFRA5353 UNCLAS
ONLY IF BLADES ARE REMOVED FROM AIRCRAFT AND REPLACED WITH LIKE
TEN).

B. SAFETY OF FLIGHT FOLLOW UP REPORT ROS CGSLD 1868, MACOMS
AND OTHER ACTION ADDRESSEES, UPON COMPLETION OF THE REQUIREMENTS SET
FORTH IN THIS MESSAGE, OR NOT LATER THAN 21 MAR 83 WILL FORWARD A
MESSAGE REPORT OF ACTION ACCOMPLISHED TO COMMANDER, TSARCOM, ATTN
DRSTS-MPSD, PER AR 95-18. UNITS WILL BE REPORTED AS DELINQUENT SEVEN
DAYS AFTER TRANSMISSION OF THIS MESSAGE. THE REPORT WILL CITE THE
SAFETY OF FLIGHT MESSAGE, THE AIRCRAFT MISSION DESIGN SERIES, AND
SERIAL NUMBERS OF THE AIRCRAFT. WHEN MAIN ROTOR BLADE SERIAL NUMBERS
THAT ARE LISTED IN PARAGRAPH 3 ARE FOUND, ALSO LIST THE BLADE SERIAL
NUMBER AND THE ACTIONS THAT WERE PERFORMED ON IT. IF ALL REQUIRED
ACTIONS ARE NOT COMPLETED, THE REPORT WILL ALSO CONTAIN THE DATE BY
WHICH IT IS ESTIMATED THEY WILL BE COMPLETED. WHEN THE ESTIMATED
DATE OF COMPLETION ARRIVES, THE SERIAL NUMBERS REPORTED PREVIOUSLY AS
NOT COMPLETED WILL BE REPORTED BY MESSAGE, WITH PRESENT STATUS
AND REVISED ESTIMATED COMPLETION DATE. REVISED ESTIMATED COMPLETION
DATES AND STATUS REPORTS WILL CONTINUE TO BE SUBMITTED UNTIL ALL
REQUIRED ACTIONS HAVE BEEN COMPLETED AND REPORTED. FOR AIRCRAFT IN
REPAIR MAINTENANCE, AWAITING DISPOSITION OR IN STORAGE, TAKE THE

AGE 24 RUCIFRA5353 UNCLAS
FOLLOWING ACTIONS

- (1) IMMEDIATELY UPDATE THE AIRCRAFT LOGBOOK BY ENTERING THE
REQUIREMENTS FOR ACCOMPLISHMENT OF THIS MESSAGE ON THE DA FORM
488-13.
- (2) REPORT BY RETURN MESSAGE BY AIRCRAFT MDS AND SERIAL
NUMBER THAT THE AIRCRAFT LOGBOOK HAS BEEN UPDATED TO INCLUDE THIS
MESSAGE REQUIREMENT.
- (3) PROVIDE A POINT OF CONTACT, WITH TELEPHONE NUMBER ON ALL
REPORTS.

3. POINT OF CONTACT FOR QUESTIONS CONCERNING FORMS OR REPORTING IS
R. RAY PEARCE, DRSTS-MPSD, AUTOVON 693-3261, COMMERCIAL
14-263-3261.

4. LOGISTIC POINT OF CONTACT IS MS. CONNIE SIMPSON, AUTOVON
93-3736, COMMERCIAL 314-263-3736.

5. FOREIGN MILITARY SALES/FMS RECIPIENTS REQUIRING CLARIFICATION OF
THIS MESSAGE SHOULD CONTACT USATSARCOM, DRSTS-I, CW4 JIM CRONIN,
AUTOVON 693-2813, COMMERCIAL 314-263-2813.

6. TSARCOM TECHNICAL POINTS OF CONTACT ARE MESSRS. JIMMY SIMON AND
D SOTEROPoulos, DRSTS-MEA, AUTOVON 693-3388, COMMERCIAL (314)
63-3388.

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